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# 弹箭发射模拟的关键问题及解决方法

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# 弹箭发射概述



# 弹箭发射仿真的物理问题

## 弹箭发射需要考虑的一些因素

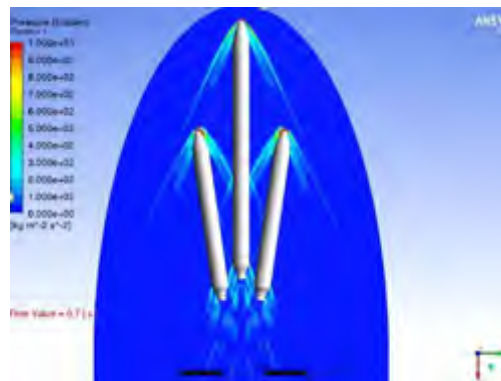
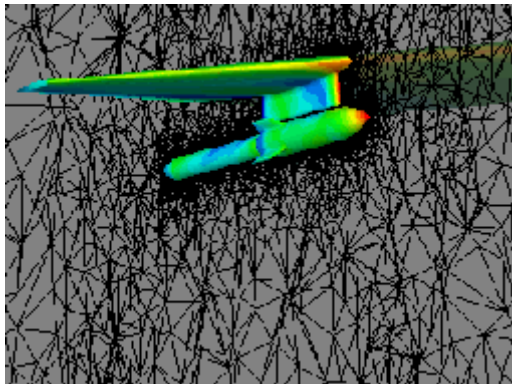
## 仿真需求

- 气动分析 → ✓ DBS+湍流模型
- 激波捕捉 → ✓ DBS+湍流模型+高精度通量分裂格式
- 气动弹性 → ✓ 1 way & 2 way FSI
- 气动加热 → ✓ Fluent共轭换热
- 弹道轨迹 → ✓ Remeshing/OverSet Mesh + 6DOF
- 噪声分析 → ✓ Fluent噪声分析模块
- ..... → ✓ ANSYS Modules

# 弹箭发射模拟的关键问题及解决方法

## □ 弹箭发射及弹道轨迹设计

- Remeshing + 6DOF
- **OverSet Mesh + 6DOF**



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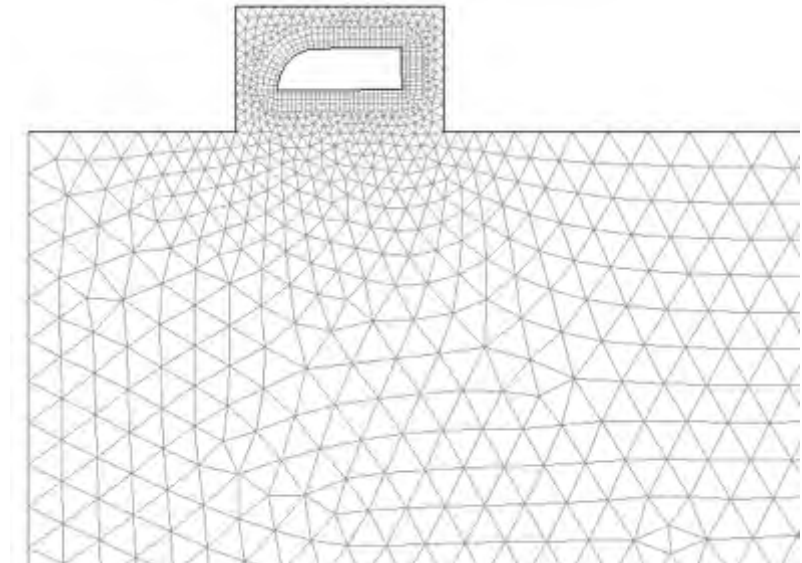
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Remeshing

# Remeshing Method - Introduction

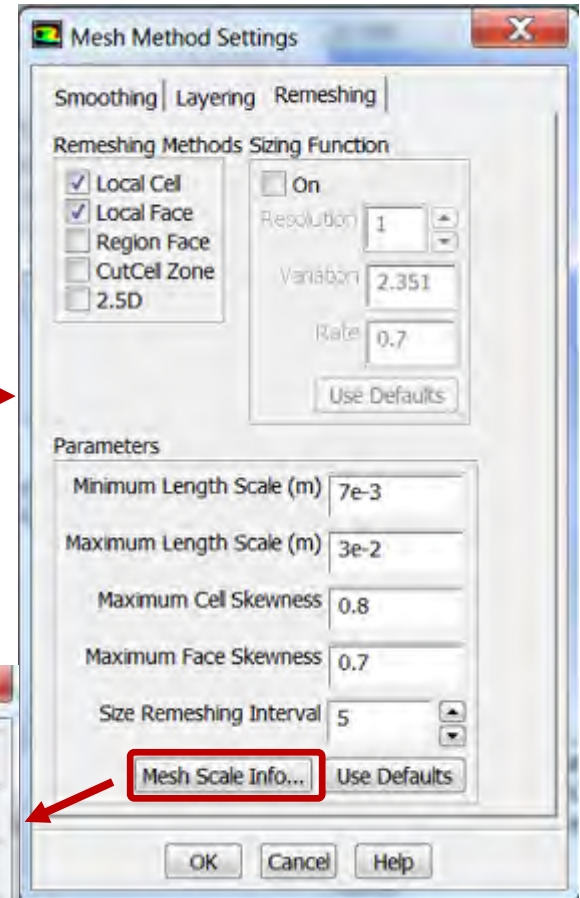
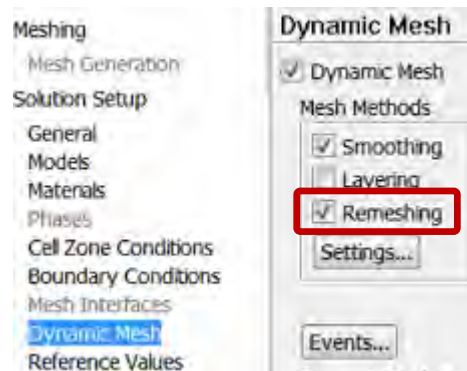
- The remeshing method makes it possible to simulate problems with large relative motion of boundaries
  - ✓ Cells and faces are remeshed when skewness or size exceeds specified limits
  - ✓ The number of nodes and their connectivity changes as cells/faces are added or deleted
- Available for:
  - ✓ tri & tet meshes (with or without prism layers)
    - (i.e. for both 2D and 3D)
  - ✓ 2.5D Prism zones (extruded tri elements)
  - ✓ Cutcell
- Remeshing and smoothing are typically used together
  - ✓ Produces better quality mesh
  - ✓ Allows larger time steps



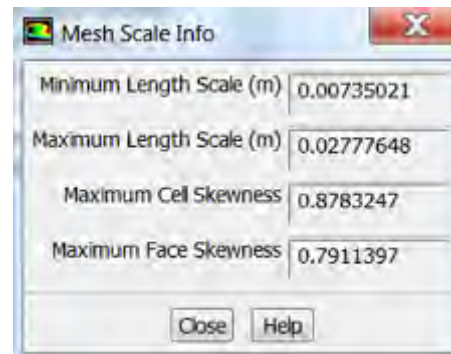
# Remeshing

- To enable the Remeshing method

- Activate Dynamic Mesh
- Select Remeshing under Mesh Methods
- Click on settings
  - Remeshing tab



- Select Method
  - Local cell/ Local Face
- View “mesh scale info”
- Select parameters
  - A good starting point for the length scales are  $0.4L_{min}$ ,  $1.4L_{max}$

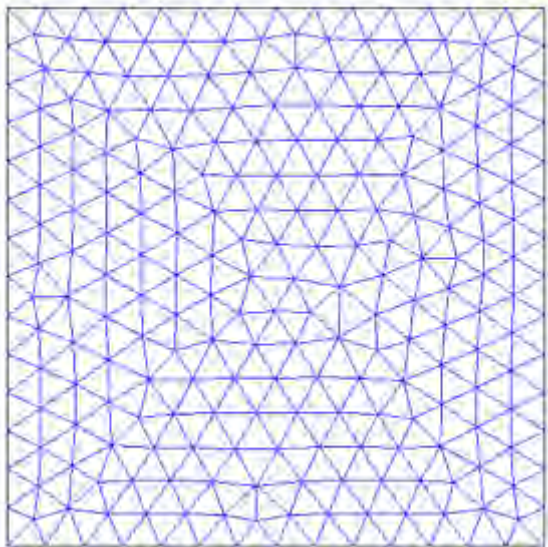


# Remeshing

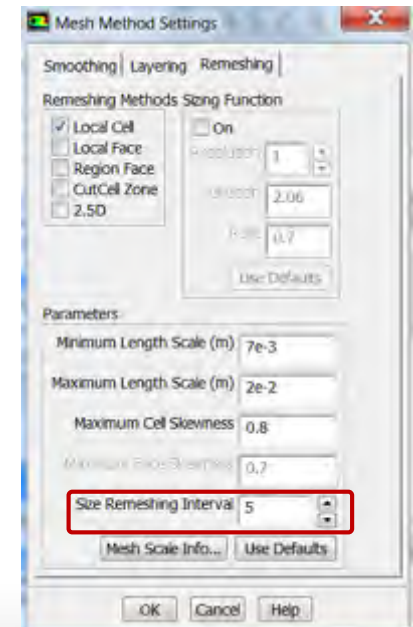
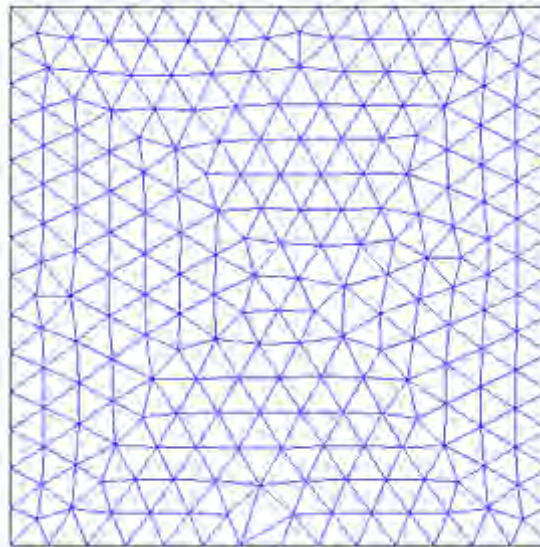
- **Size Remeshing Interval**

- Size Remesh Interval (SRI) controls how frequently cells are marked for remeshing based on the cell size.
  - If SRI is large, then remeshing interval is dominated by skewness.
  - If SRI is small, then remeshing interval is strongly affected by both cell size and skewness.
  - Setting SRI to 1 is often necessary for larger time steps
    - It is common that the time step size is limited by the remeshing rather than the flow solution

Size Remeshing Interval = 10



Size Remeshing Interval = 1





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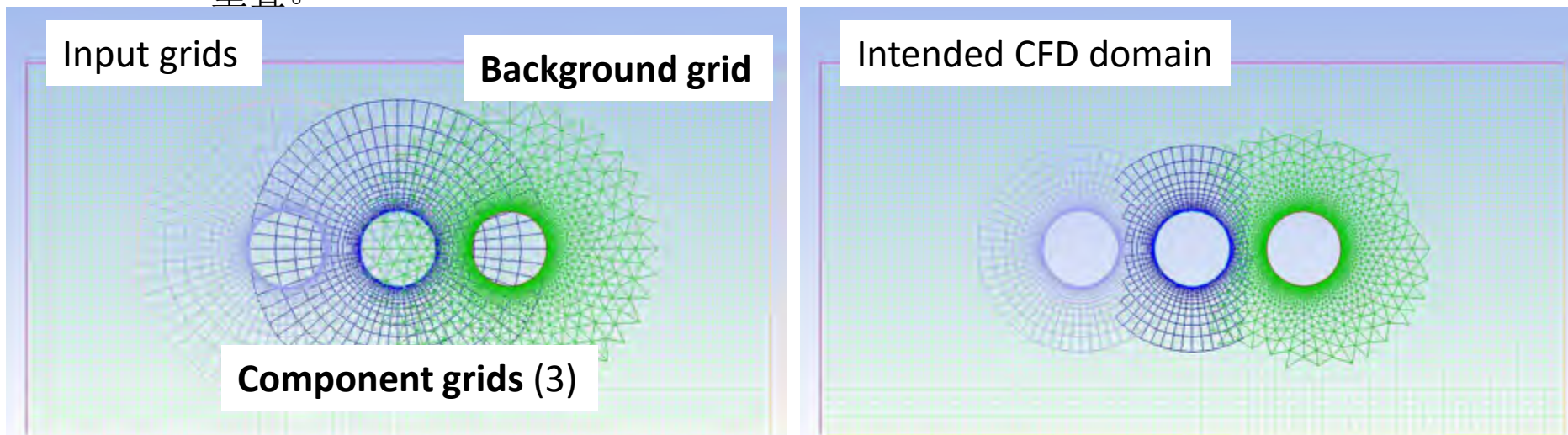
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# Overset Meshing

# 重叠网格概述

- 由部件网格构建计算域的新方法
  - 单域或共节点 连接多区域网格
  - 匹配面之间的网格连接是通过 $non-conformal$ 来实现的。  
网格之间的连通性是建立在面level上的。
  - 通过overset interface连接重叠网格单元区域
    - 单独划分Parts网格，并将其嵌入到背景网格中
    - 重叠区域的连通是通过网格单元数据插值来实现的，网格之间需要有足够的重叠。

旧的现有方式



# 重叠网格特点

- **ANSYS Fluent功能扩展**

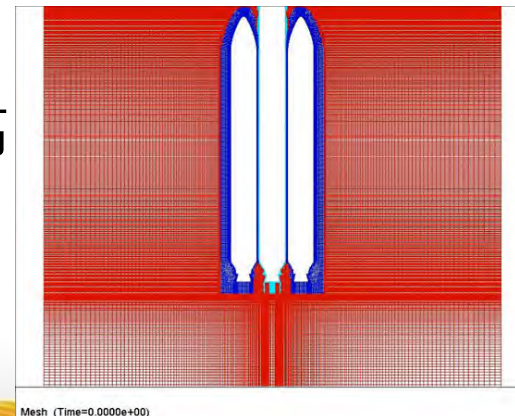
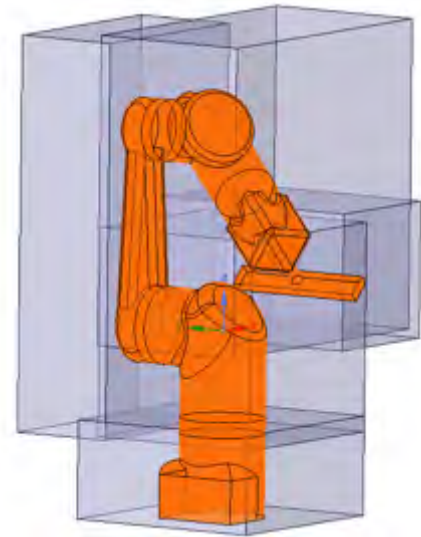
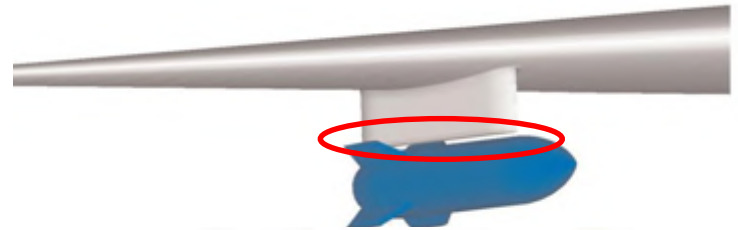
- 克服了动网格的一些限制
  - 可以处理具有小缝隙部件的相对网格运动

- **易用**

- 简化了复杂几何的网格生成
- 避免了动网格应用中网格重构失败和动网格设置的一些问题
- 更容易的构型变化和组件交换

- **求解质量**

- 重叠网格在网格运动期间始终可以保持很高的网格质量
- 局部结构网格在非结构网格中的使用



# 重叠网格构建条件

## • 背景网格

- 网格单元区域内没有重叠类型边界，背景网格必须是共节点的网格

## • 组件网格

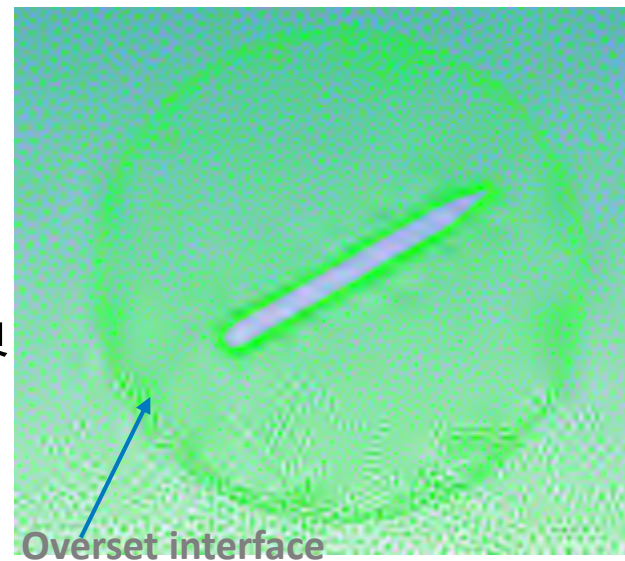
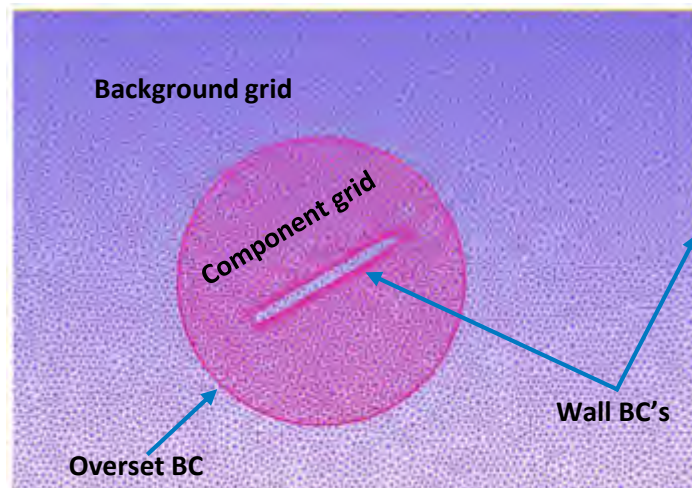
- 网格单元区域至少有一个重叠类型边界，组件网格一定是共节点网格

## • 重叠边界

- 新的特定的重叠边界条件类型
- 将组件与其他网格通信的区域指定为重叠边界

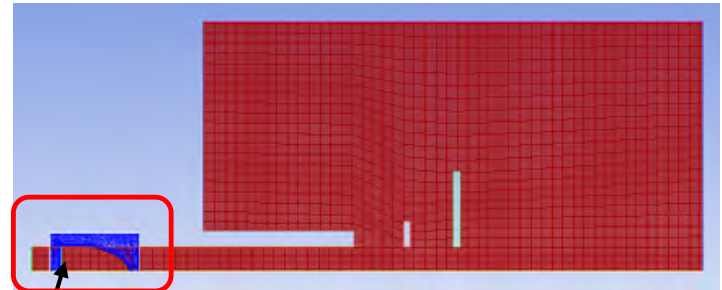
## • 重叠交界面

- 组件网格和背景网格是成对出现的
- 至少需要一个背景网格和一个组件网格

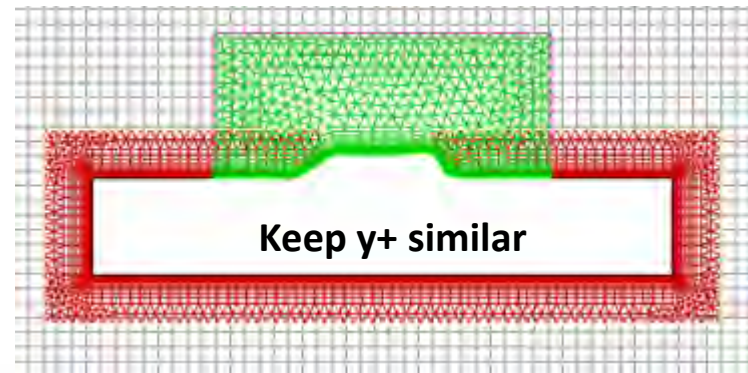
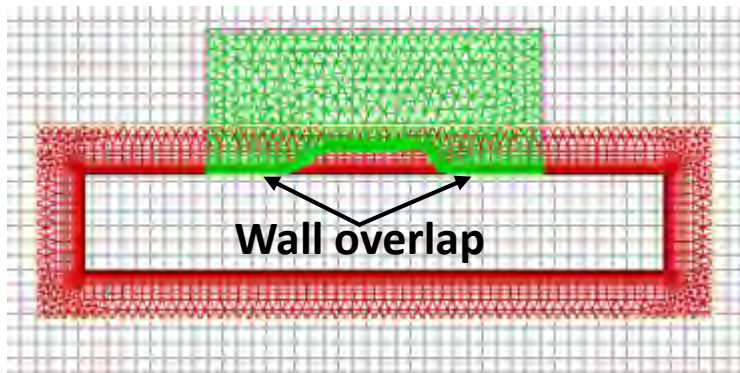


# 重叠拓扑

- **组件网格可任意重叠**
  - 重叠网格边界是允许的
- **允许边界重叠，但不允许边界交叉**
  - 挖洞过程中自动进行匹配检测/相同边界条件类型的重叠几何 (wall, symmetry, ...)
- **交叉壁面边界需要额外 衣领 网格**
  - 衣领网格连接交叉几何
    - 生成边界重叠网格，并去除交叉
  - 生成了孤点 ( 没用衣领网格 )

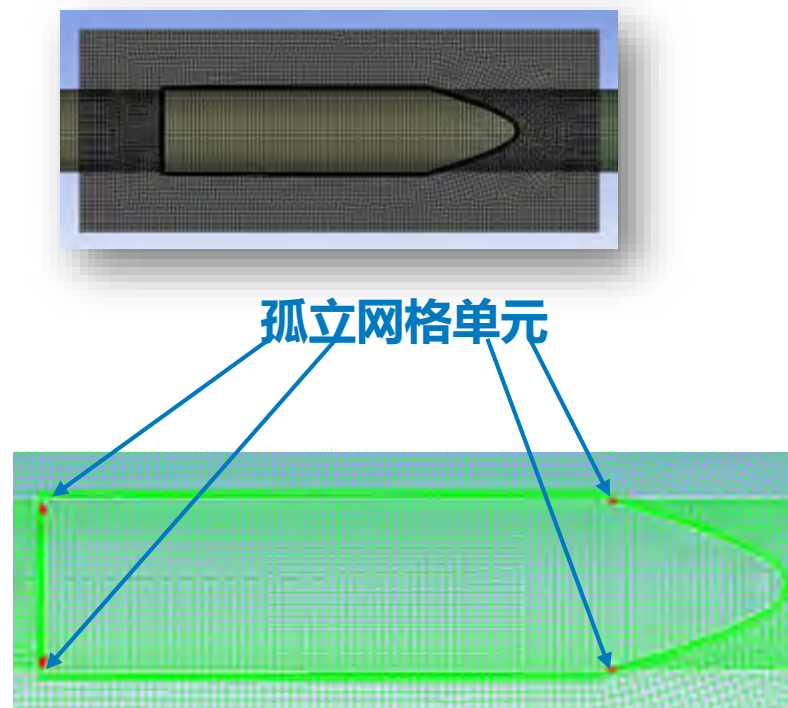
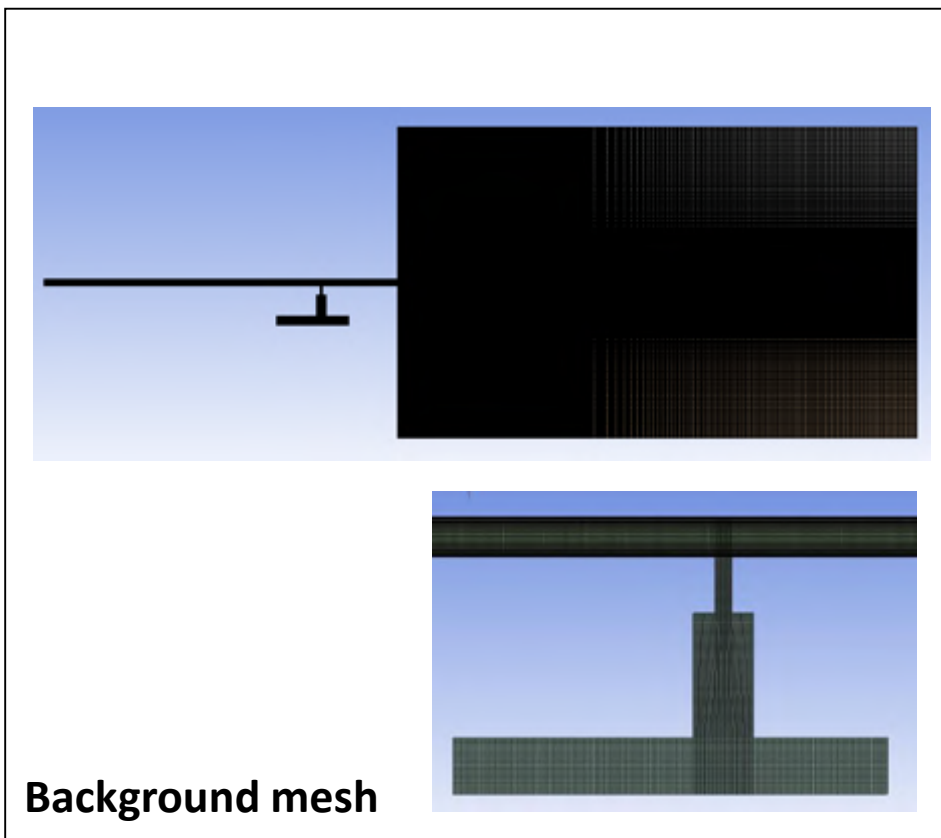


Wall intersections  
not permissible



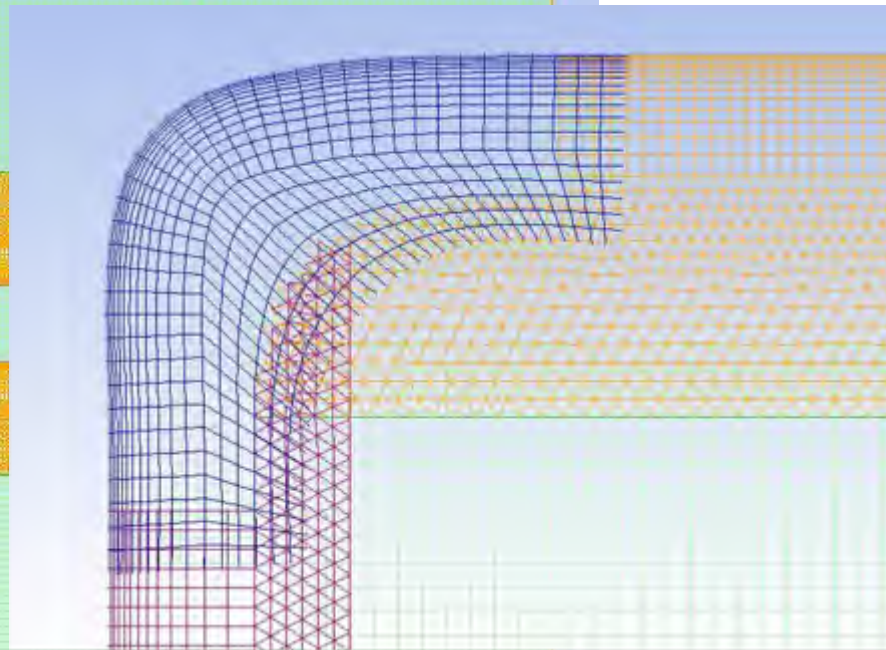
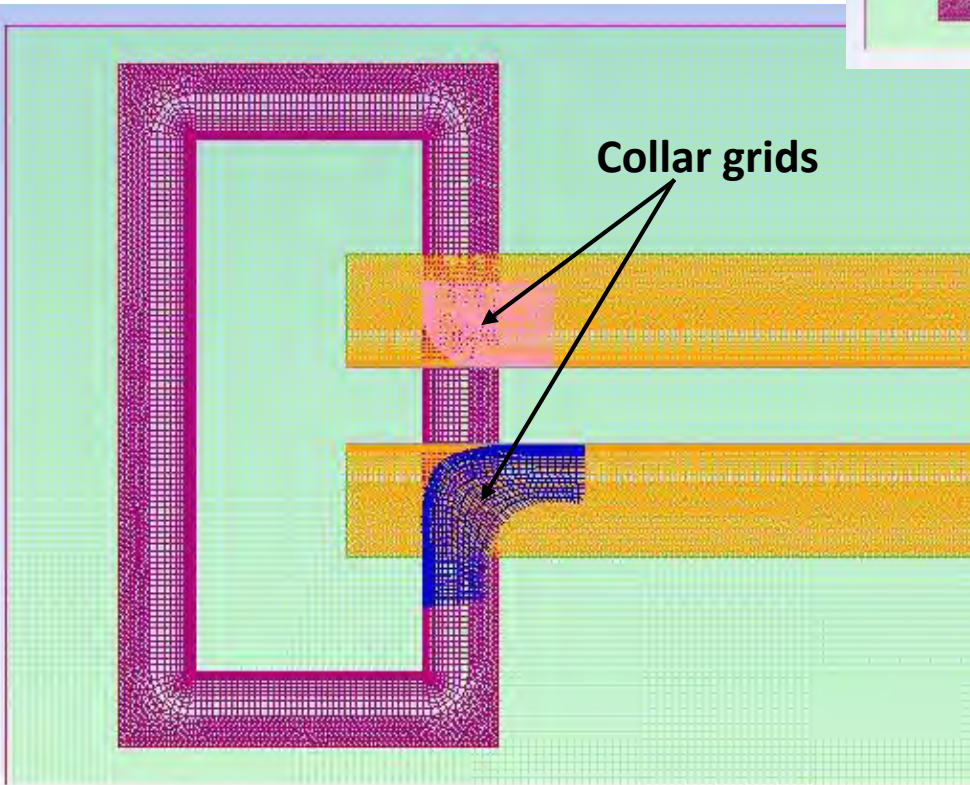
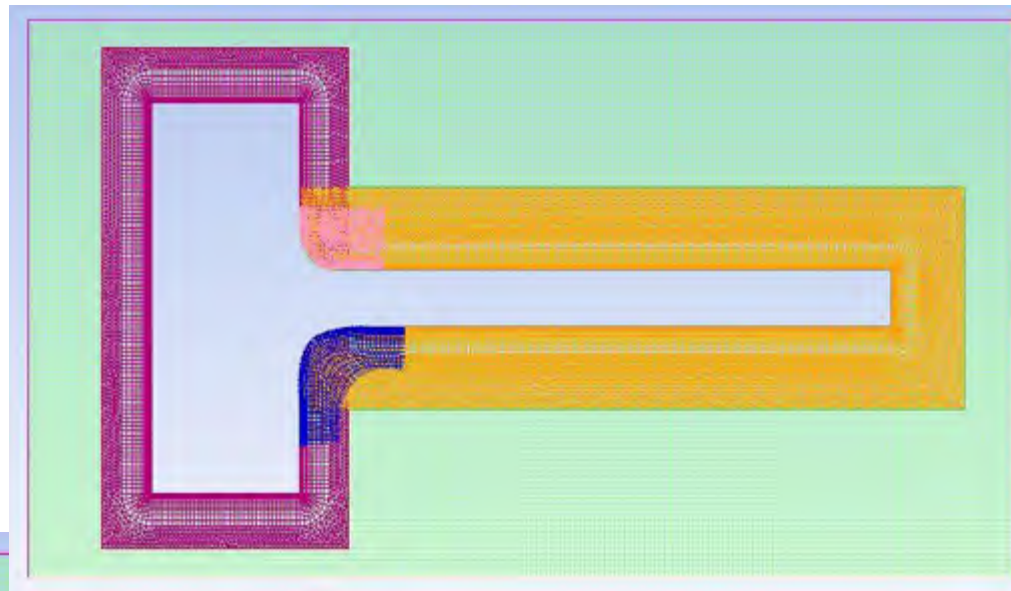
# 网格拓扑

- 没用衣领网格处理的交叉壁面边界



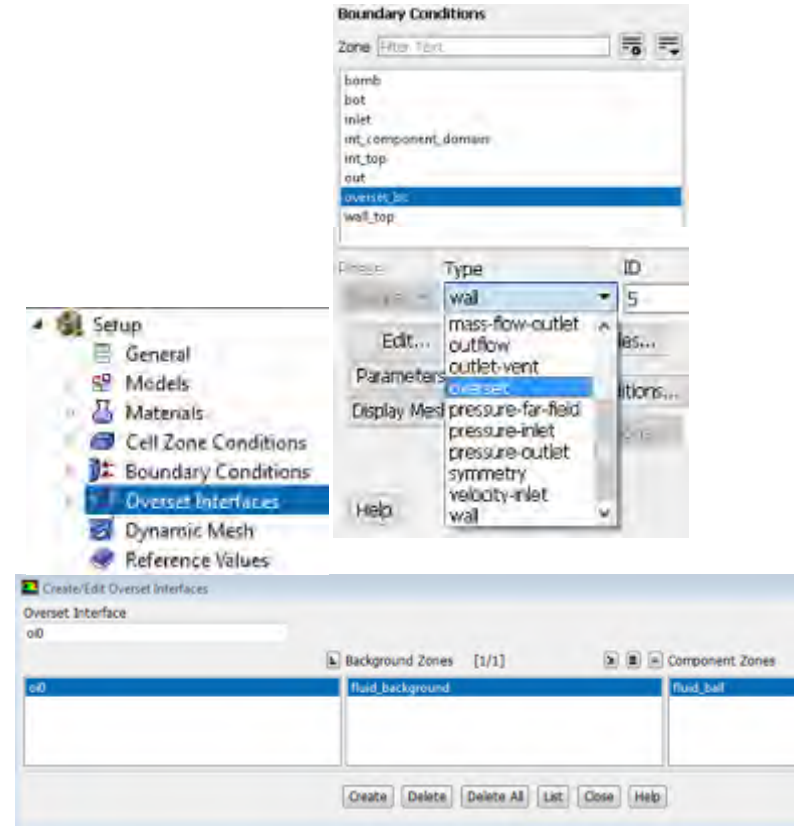
# 重叠拓扑

- 用衣领网格处理交叉壁面边界
  - 消除交叉，并用重叠壁面取代



# 重叠网格设置

- Read all related meshes into fluent
- Set the overset boundary condition
- Create the interface
  - ✓ Select background and component meshes
- Define grid priorities if needed
  - ✓ *define/overset-interfaces/grid-priorities*
- Initialize to intersect the interface
  - ✓ Or use TUI command to intersect the interface without initializing the flow field (activate expert options first)  
*define/overset-interfaces/intersect*
- An overset interface will automatically be created during initialization if an overset BC is defined but no interface



注：User interface 与 non-conformal interfaces 类似, 用最小的用户输入就可以处理所有支持的拓扑

- 不需要连接信息
- 自动检测重叠区域



# 重叠网格支持的选项&模型 @ R18.0

- 压力基耦合求解器 (Planar 2D & 3D)
- 密度基求解器 (Planar 2D & 3D)
- Laminar, standard k-epsilon & standard k-omega , k-omega SST
- Compressible flows, heat transfer
- Volume of Fluid (VOF)
- Dynamic and sliding meshes with the first-order transient formulation
- 与Fluent支持的所有网格单元和类型都兼容  
包括 : *polyhedra, hexcore and cutcell meshes*
- 与网格自适应兼容

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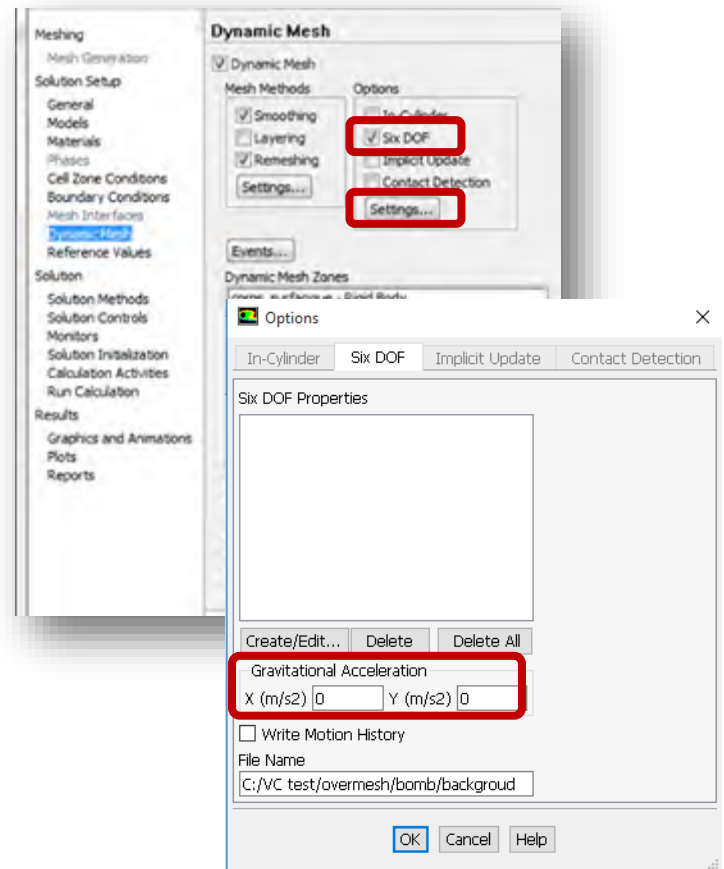
**Coupled 6 DOF (Degree of freedom)**

# Introduction

- **The motion of the parts have been explicitly defined**
- **The motion is known at the start of the simulation,**
  - Due perhaps, to parts that are mechanically driven
- **However in some cases the motion of the body is driven by the forces exerted by the fluid**
  - Therefore the motion is dependent on the flow solution
  - This can be solved in Fluent using the "6 Degree of Freedom" option
- **The 6 DOF solver computes the trajectory of the object with respect of the**
  - Aerodynamic forces/moments calculated
  - Additional external or internal forces
    - **Gravity**
    - **Thrust**
    - **Ejector**
- ***Motion and flow field are thus coupled → coupled motion***

# 6DOF – Principle

- **6DOF Parameter settings are done in 2 steps**
  - Dynamic Mesh panel
    - **Provides Gravitational forces**
    - **Initial conditions**
      - CG location & orientation
      - CG linear & angular velocity
  - UDF/GUI
    - **Mass**
    - **Momentum of inertia , ...**
- **Fluent computes**
  - Aerodynamic forces and moments + New CG location and orientation
    - **The translational acceleration is computed from the overall force balance**
      - integrate to calculate the translational velocity.
    - **The angular acceleration is computed from the overall moment balance**
      - integrate to calculate the angular velocity



# 6DOF - UDF

- To specify custom properties to a moving object, use the following macro
  - DEFINE\_SDOF\_PROPERTIES* (names, properties, dt, time, dtime)

SDOF_MASS	/* mass */	SDOF_LOAD_LOCAL,	/* boolean */
SDOF_IXX,	/* moment of inertia */	SDOF_LOAD_F_X,	/* external force */
SDOF_IYY,	/* moment of inertia */	SDOF_LOAD_F_Y,	/* external force */
SDOF_IZZ,	/* moment of inertia */	SDOF_LOAD_F_Z,	/* external force */
SDOF_IXY,	/* product of inertia */	SDOF_LOAD_M_X,	/* external moment */
SDOF_IXZ,	/* product of inertia */	SDOF_LOAD_M_Y,	/* external moment */
SDOF_IYZ,	/* product of inertia */	SDOF_LOAD_M_Z,	/* external moment */

SDOF_ZERO_TRANS_X,	/* boolean, suppress translation in x-direction */
SDOF_ZERO_TRANS_Y,	/* boolean, suppress translation in y-direction */
SDOF_ZERO_TRANS_Z,	/* boolean, suppress translation in z-direction */

SDOF_ZERO_ROT_X,	/* boolean, suppress rotation around x-axis */
SDOF_ZERO_ROT_Y,	/* boolean, suppress rotation around y-axis */
SDOF_ZERO_ROT_Z,	/* boolean, suppress rotation around z-axis */

SDOF_SYMMETRY_X,	/* normal vector of symmetry plane for half model */
SDOF_SYMMETRY_Y,	/* normal vector of symmetry plane for half model */
SDOF_SYMMETRY_Z,	/* normal vector of symmetry plane for half model */

# 6DOF – UDF example

- 6DOF UDF example case with ejector forces & momentum included

```
#include "udf.h"

DEFINE_SDOF_PROPERTIES(delta_missile, prop, dt, time, dtime)
{
    real y;

    prop[SDOF_MASS] = 907.185;
    prop[SDOF_IXX] = 27.116;
    prop[SDOF_IYY] = 488.094;
    prop[SDOF_IZZ] = 488.094;
    /* add injector forces, moments */
    {
        y = CURRENT_TIME;
        if (y <= 0.15)
        {
            prop[SDOF_LOAD_F_Z] = 10676.0;
            prop[SDOF_LOAD_M_Y] = -1920.0;
        }
        else
        {
            prop[SDOF_LOAD_F_Z] = 0;
            prop[SDOF_LOAD_M_Y] = 0;
        }
    }
    printf ("\ndelta_missile: updated 6DOF properties");
}
```

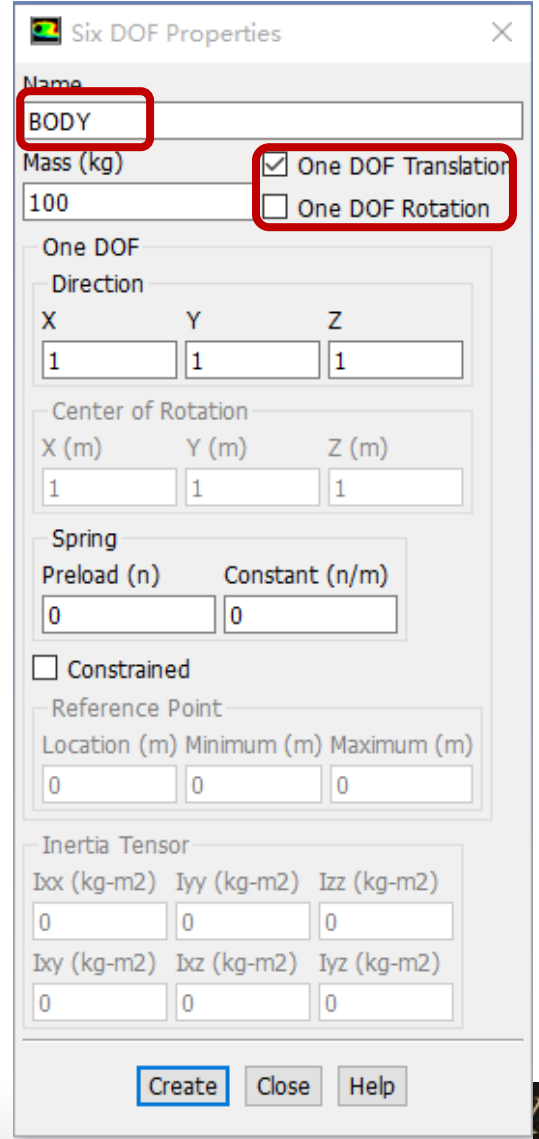
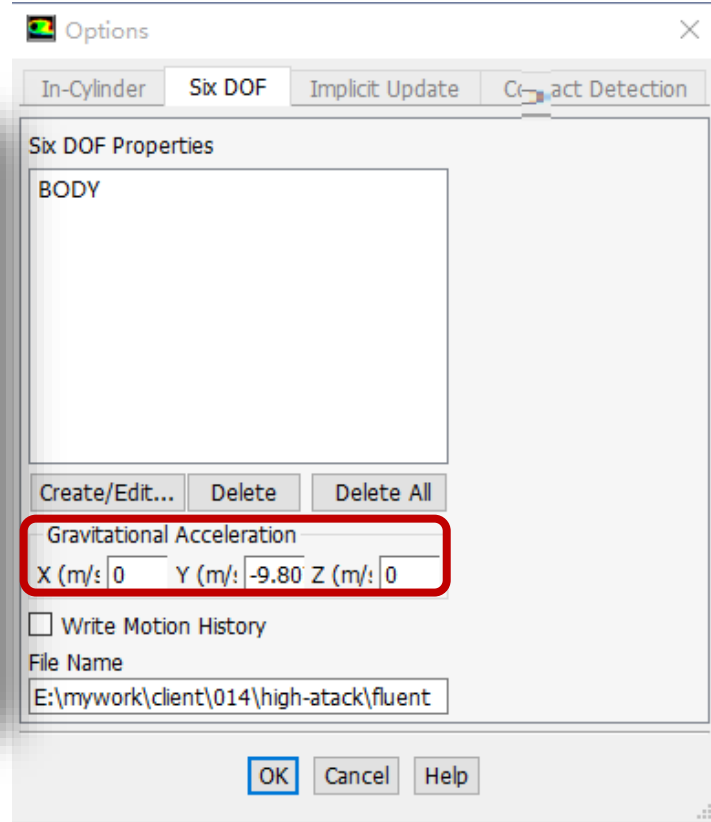
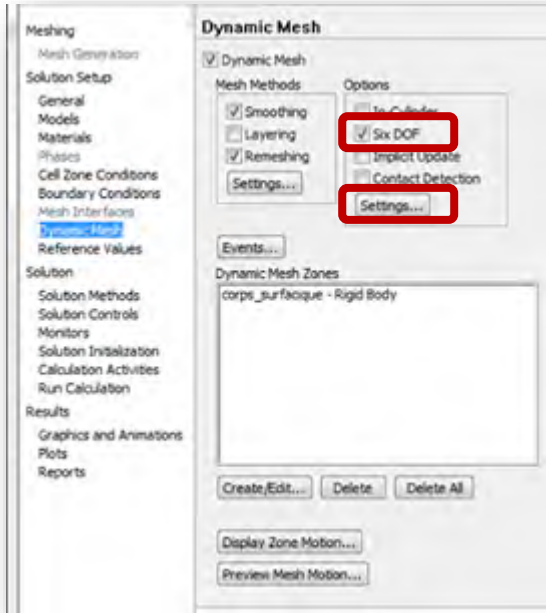
Properties of the delta\_missile

forces & moments depend on the current time

Additional forces & moments load

Removal of forces & moments load

# 6DOF – GUI



# 6DOF – GUI

### Six DOF Properties

Name: **BODY**

Moment of Inertia (kg-m<sup>2</sup>): 0

One DOF Translation  
 One DOF Rotation

One DOF Axis: X=1, Y=1, Z=1

Center of Rotation: X(m)=1, Y(m)=1, Z(m)=1

Spring: Preload (n-m)=0, Constant (n-m/rad)=0

Constrained

Reference Angle: Value (deg)=0, Minimum (deg)=0, Maximum (deg)=0

Inertia Tensor: Ixx=0, Iyy=0, Izz=0, Ixy=0, Ixz=0, Iyz=0

Buttons: Create, **Close**, Help

### Dynamic Mesh Zones

Zone Names: bwall

Dynamic Mesh Zones: [Empty List]

Type:  Rigid Body

Stationary  
 Rigid Body  
 Deforming  
 User-Defined  
 System Coupling

Buttons: Motion Attributes, Geometry Definition, Meshing Options, Solver Options

**Six DOF UDF/Properties**: BODY

Relative Motion:  On,  On

Relative Zone: [Dropdown]

Six DOF:  On,  Passive

Center of Gravity Location: X(m)=0, Y(m)=0, Z(m)=0

Rigid Body Orientation: Theta(deg)=0, Axis\_X=0, Axis\_Y=0, Axis\_Z=0

Center of Gravity Velocity: V\_X(m/s)=0, V\_Y(m/s)=0, V\_Z(m/s)=0

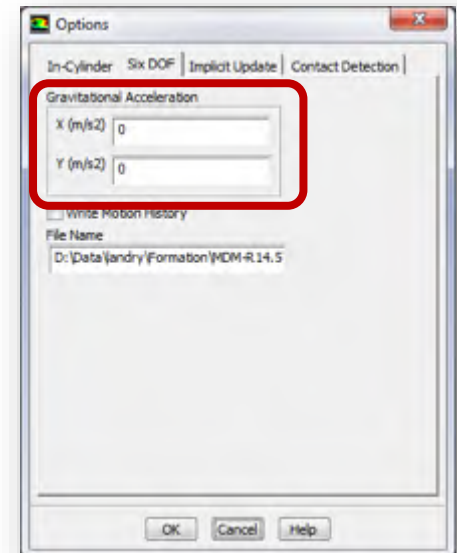
Rigid Body Angular Velocity: Omega\_X(rad/s)=0, Omega\_Y(rad/s)=0, Omega\_Z(rad/s)=0

Buttons: Create, Draw, Delete All, Delete, Close, Help



# Workflow for a 6DOF case

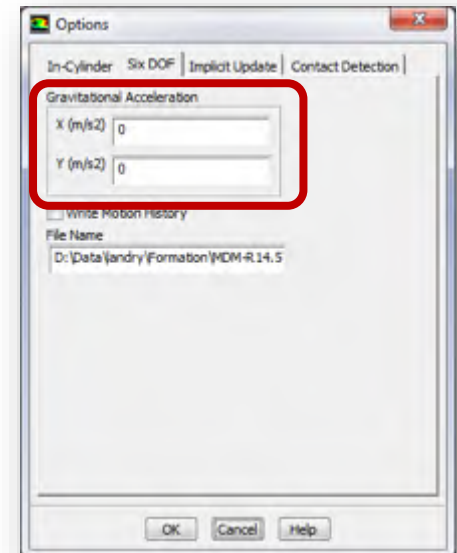
- Write the 6DOF UDF
  - Read the mesh,
  - DEFINE\_SDOF\_PROPERTIES (names, properties, dt, time, dtime)
  - Set up the flow case
- Compile & Load the UDF
  - Run till convergence (if required)
- Read the mesh,
- Switch to transient solver
  - Run till convergence (if required)
- Choose the right Dynamic mesh method(s)
- Switch to transient solver
- Switch ON the 6 DOF
  - Choose the right Dynamic mesh method(s)
- Switch ON the 6 DOF
  - Settings
    - defined Rigid properties in Fluent 6DOF GUI
    - Set the gravitational acceleration
    - define Rigid properties
    - Set the appropriate Dynamic Mesh Zones
- Save case file
- Preview mesh motion



Even if it is a coupled motion (see tips later)

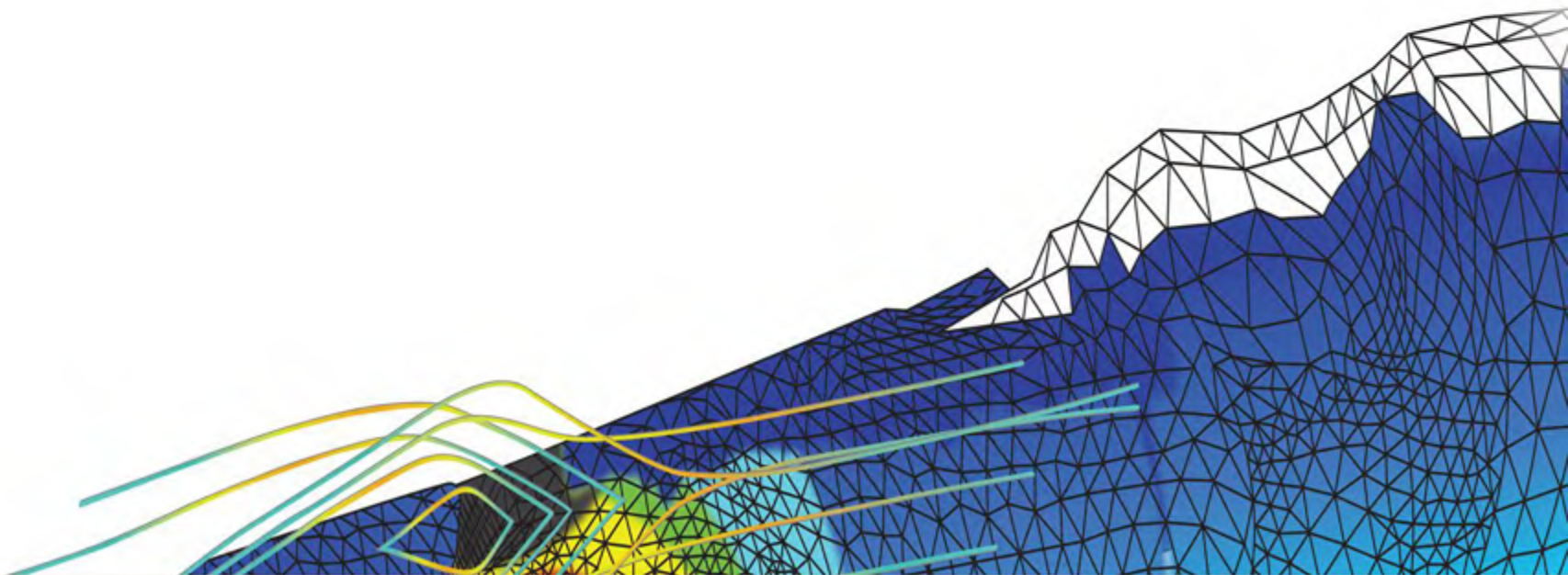
# Workflow for a 6DOF case

- **Read the mesh,**
  - Create overset mesh compute domain
  - Set up the flow case
    - **Run till convergence (if required)**
- **Switch to transient solver**
- **Switch ON the 6 DOF**
  - Settings
    - defined Rigid properties in Fluent 6DOF GUI
    - define Rigid Move attributes in dynamic zones
- **Save case file**
- **Preview mesh motion**



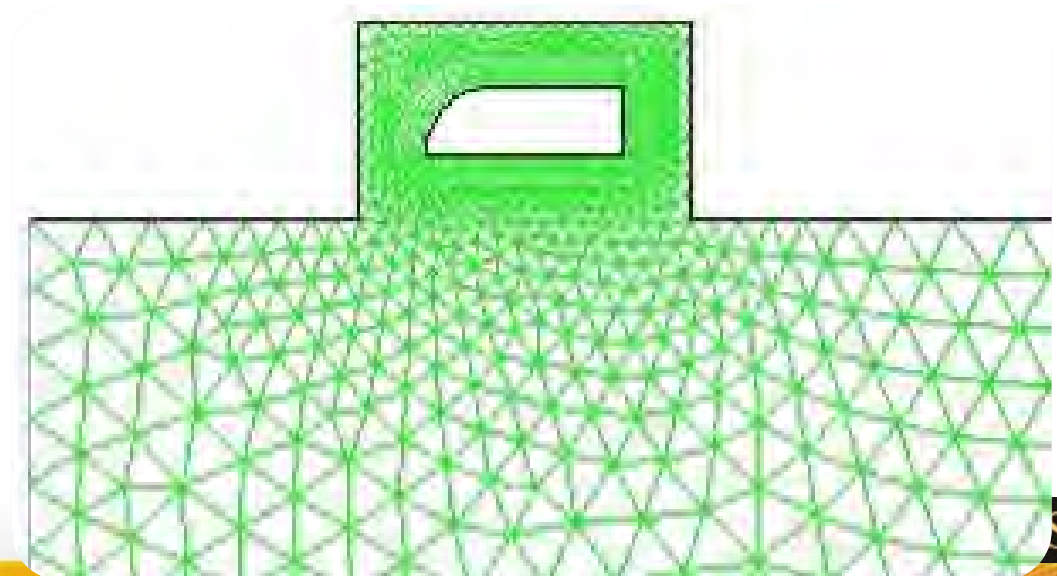
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# 应用案例



# Case 1 – Airdrop

- The object is not in contact with any external boundary
- The fluid domain is composed of 2 fluid zones
  - One prism cells
    - Moves with the falling object (to better follow up forces &  $y^+$ )
    - Define as a rigid body
  - One triangle
    - Does correspond to the deforming zone
      - Smoothing method
      - Remeshing method
        - Local cell



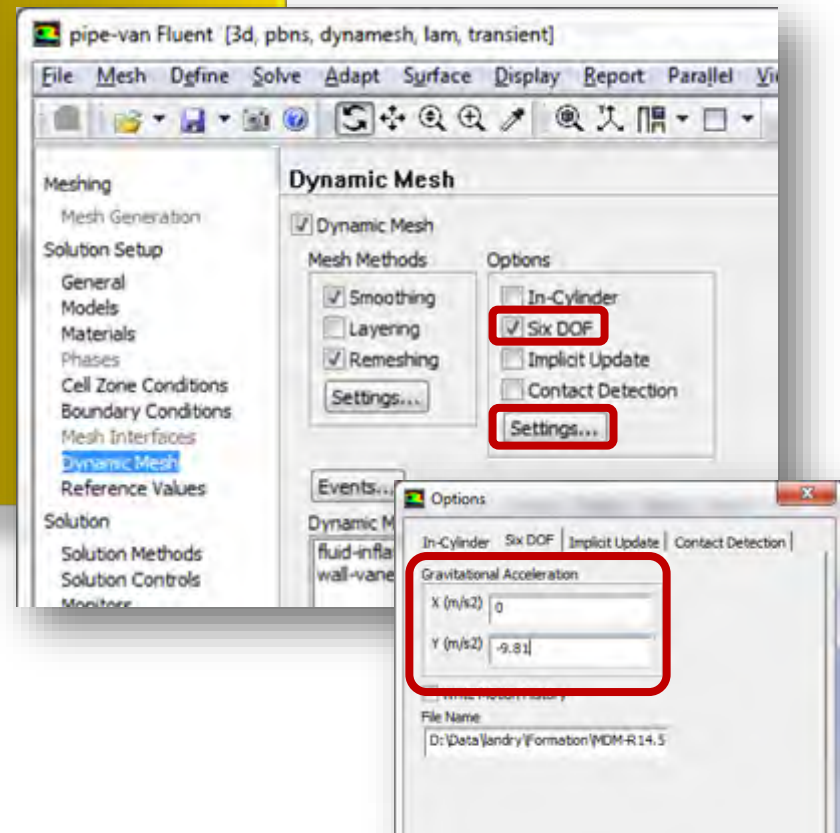
# Case 1 – Airdrop

- The 6DOF UDF

```

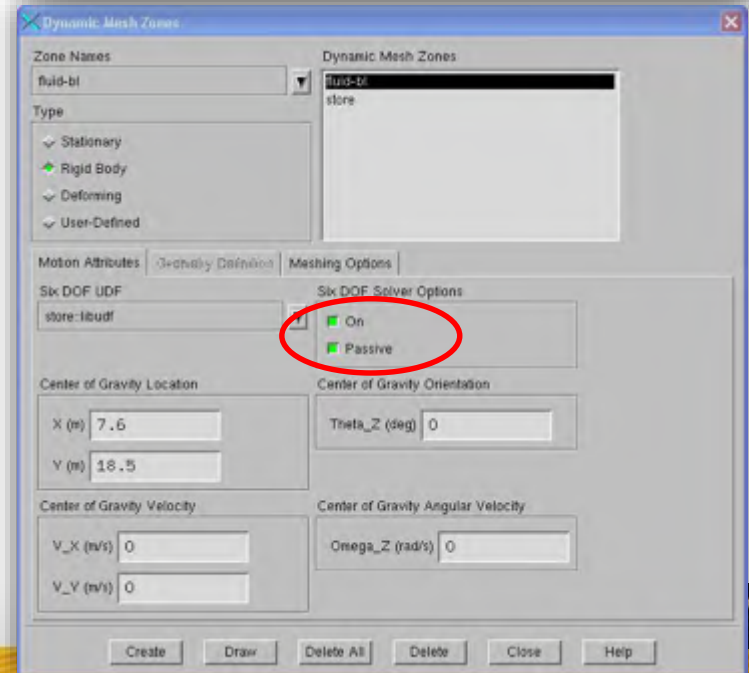
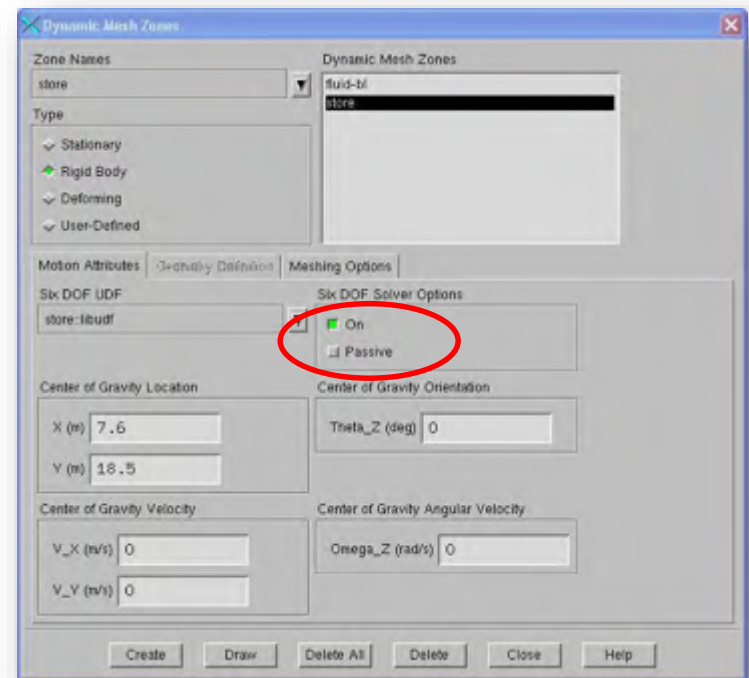
#include "udf.h"
DEFINE_SDOF_PROPERTIES(store, prop, dt, time, dtime)
{
  /* Define the mass matrix */
  prop[SDOF_MASS] = 5000.0;
  prop[SDOF_IZZ] = 5e3;
  /* add ejector forces, moments */
  if (time <= 0.3)
  {
    prop[SDOF_LOAD_F_X] = -40000;
    prop[SDOF_LOAD_F_Y] = -80000;
    prop[SDOF_LOAD_M_Z] = -2200.0;
  }
  Message0("\nupdated 6DOF properties\n");
}
  
```

- Define the gravitational acceleration



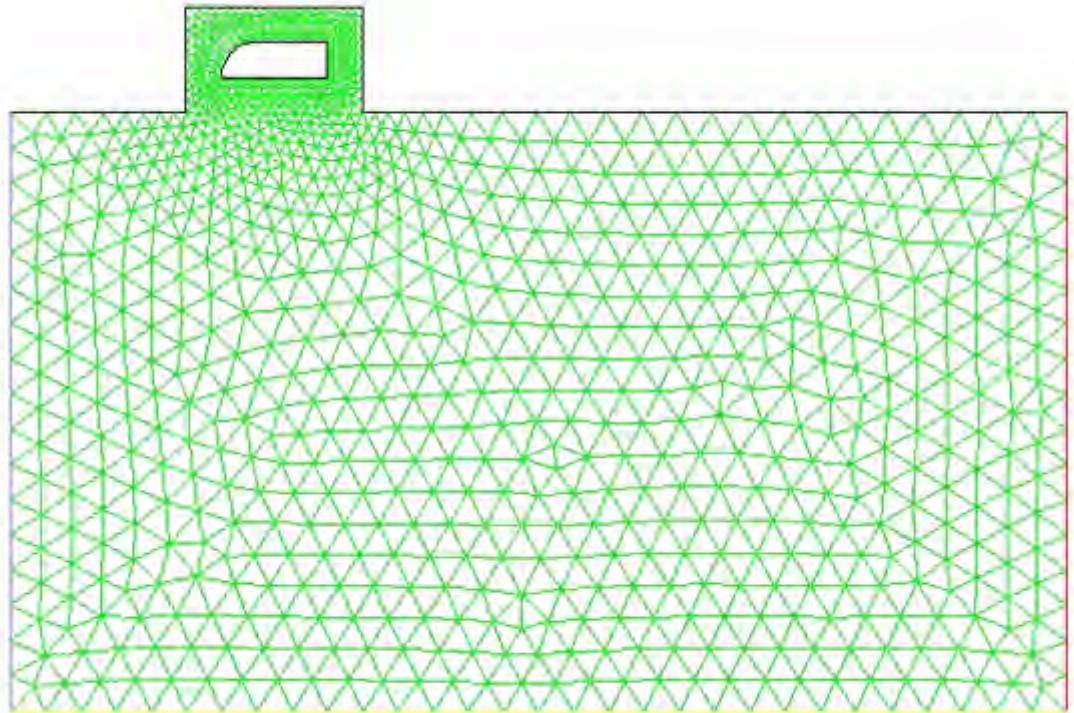
# Case 1 – Airdrop

- 2 Dynamic Mesh Zone to define
  - The wall (store)
    - Rigid body
    - Six DOF UDF + option **ON**
    - Center of gravity location & orientation
    - ...
  
  - The prism cells (fluid-bl)
    - Rigid body
    - Six DOF UDF
      - Option **ON**
        - Insure store displacement
      - **Passive** activated
        - To not take flow efforts into count
    - Center of gravity location & orientation
    - ...



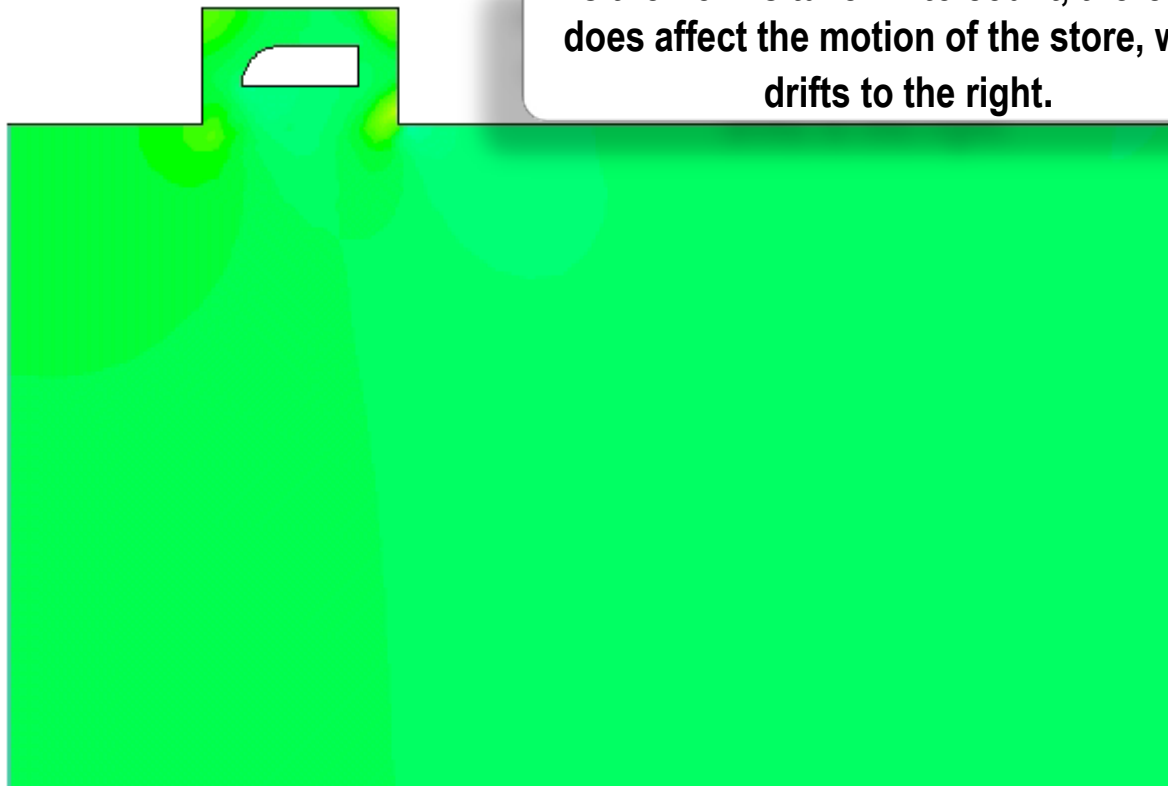
# Case 1 – Mesh Motion

- **Preview mesh motion**
  - Wise to preview the mesh motion before performing the calculations
  - Start without initializing,
- **Object simply drops under the influence of gravity and/or external forces**



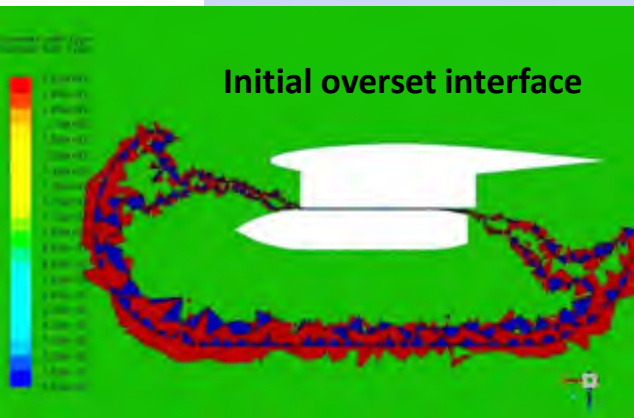
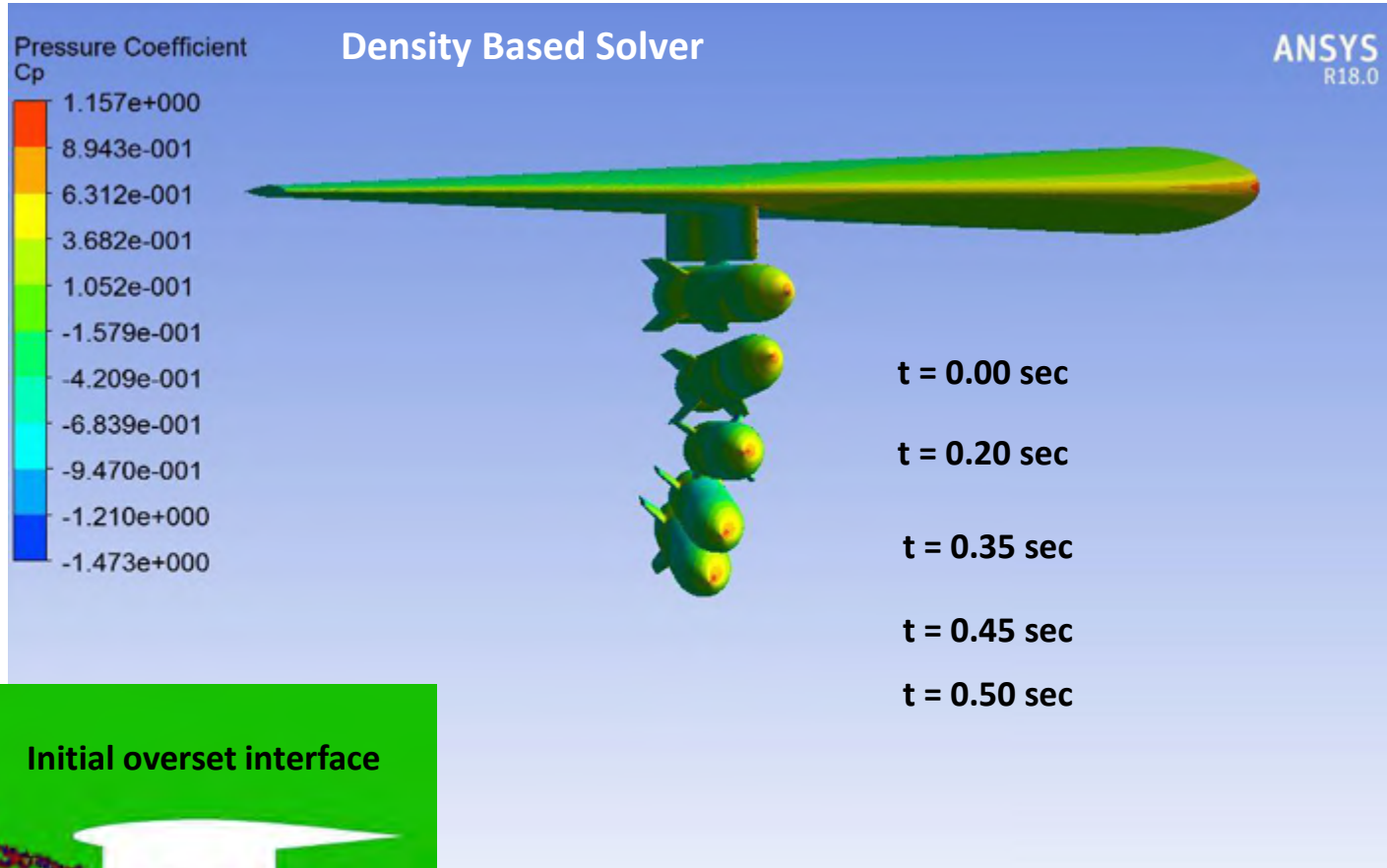
# Case 1 – Animation

- **Figure shows pressure contours**
  - Freestream Mach number of 0.8
- **The store goes forward at the start due to ejector force**

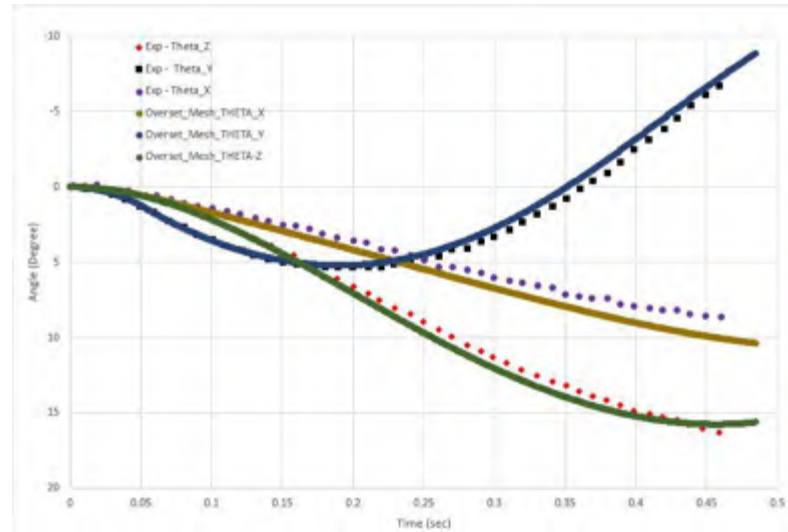
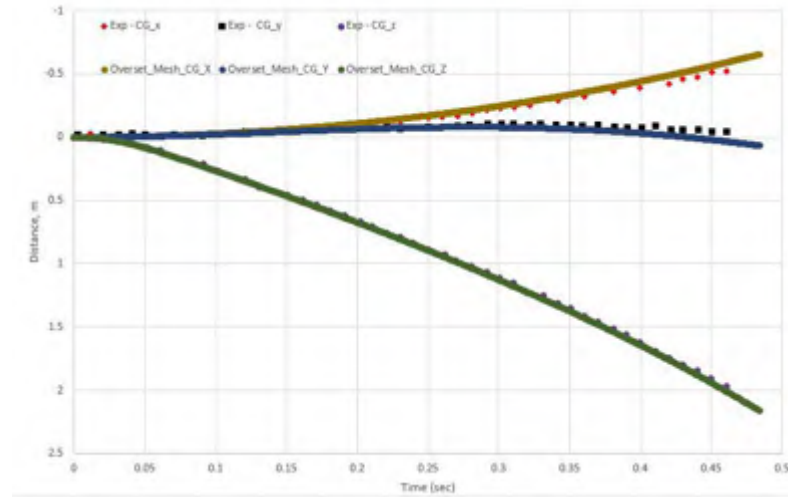
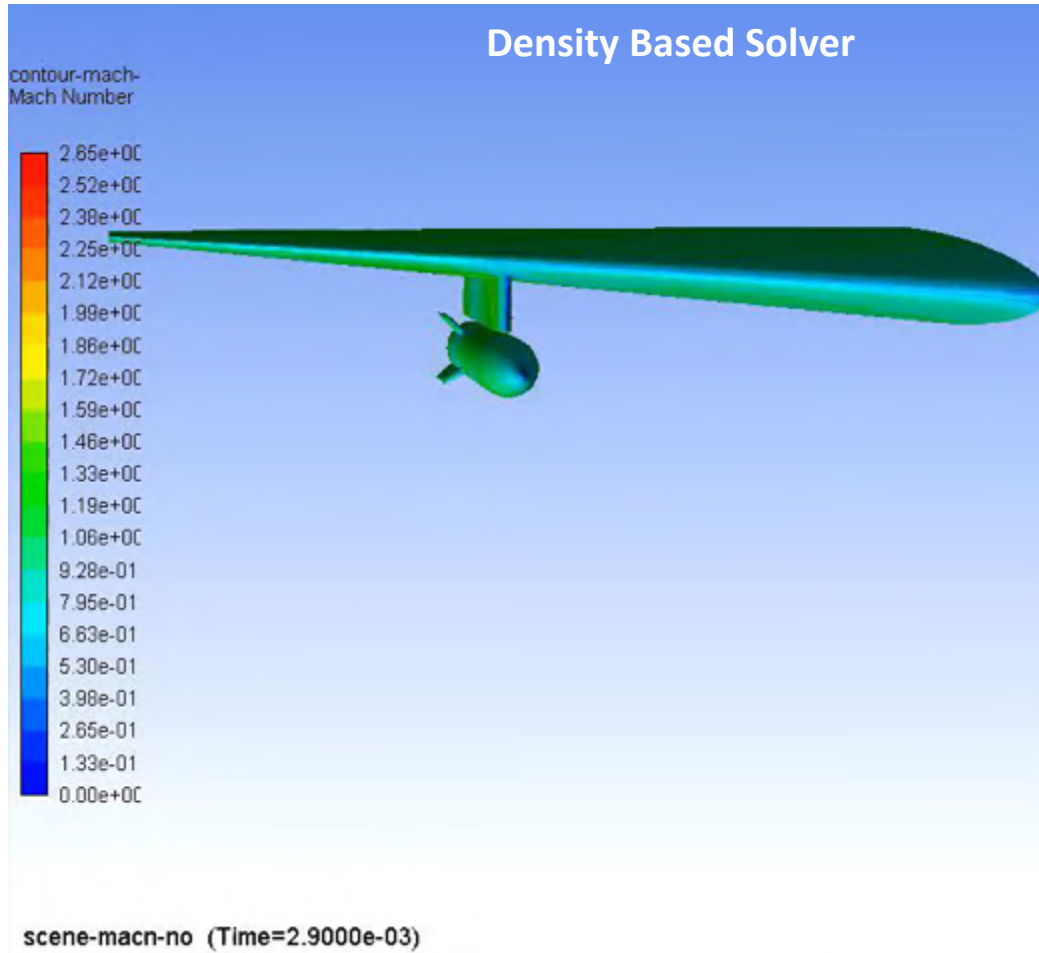




# Case2 : 跨音速投放数值模拟 & 密度基求解器



# Case2 : Result



## Case3 : 2D弹丸出膛分析

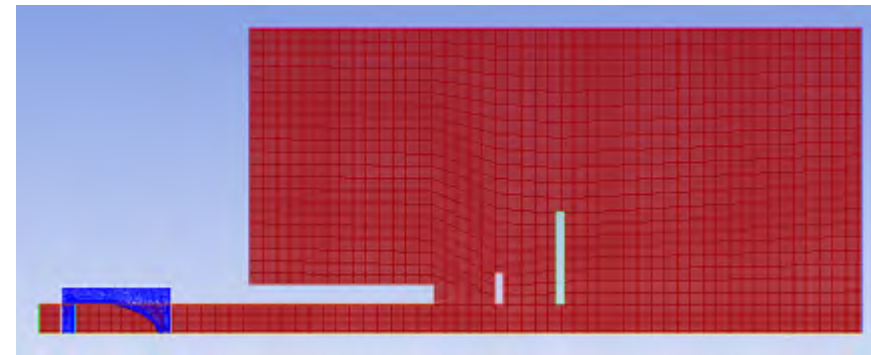
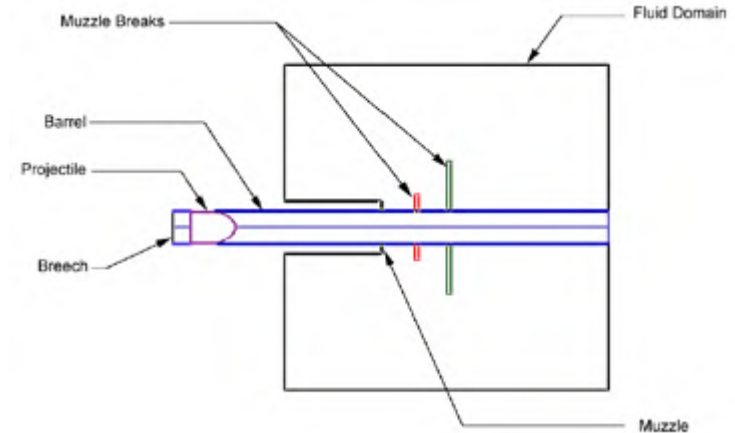
### □ 网格 & 设置

- 与层铺动网格 ( non-conformal interface ) 计算结果比较

- Pressure based coupled solver
- SSTk-omega
- Noble-Abel real gas

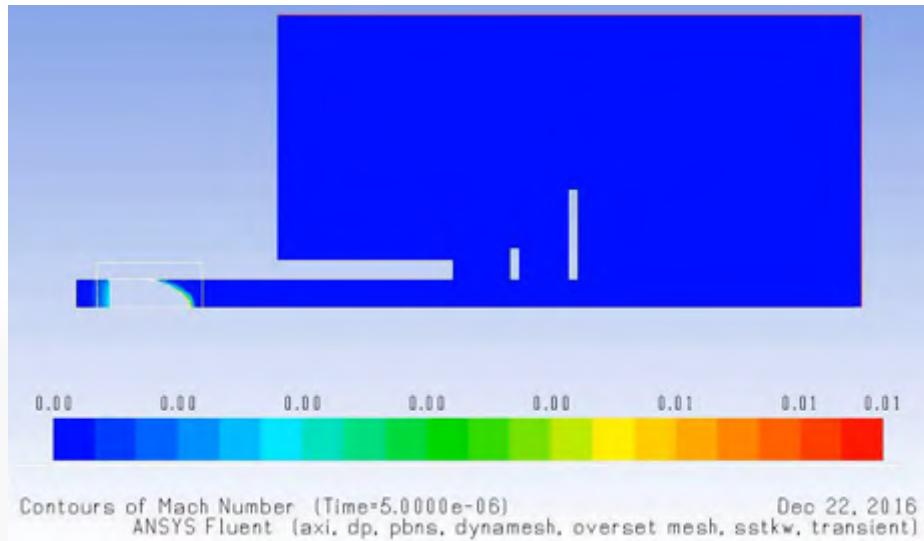
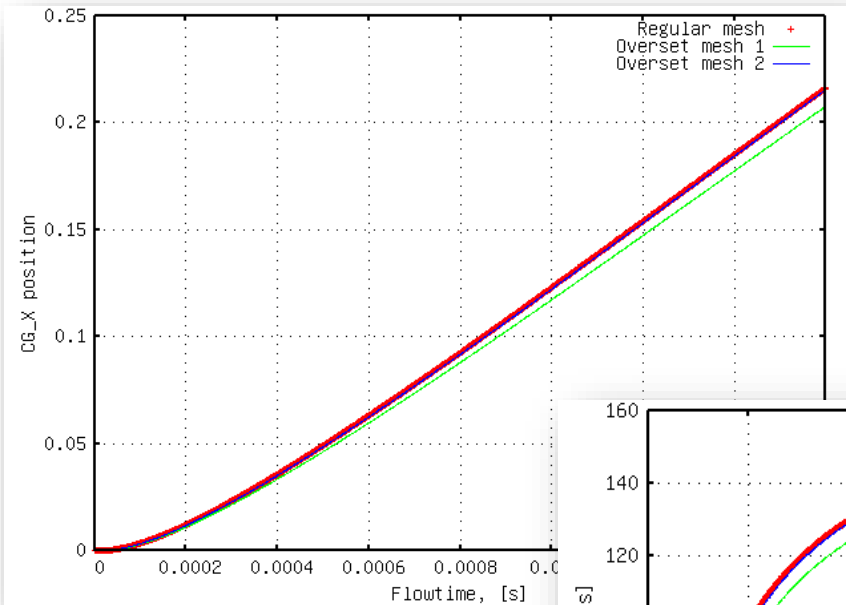
- 弹丸后的初始条件为:

$p=3000 \text{ atm}$ , and  $T=2700\text{K}$

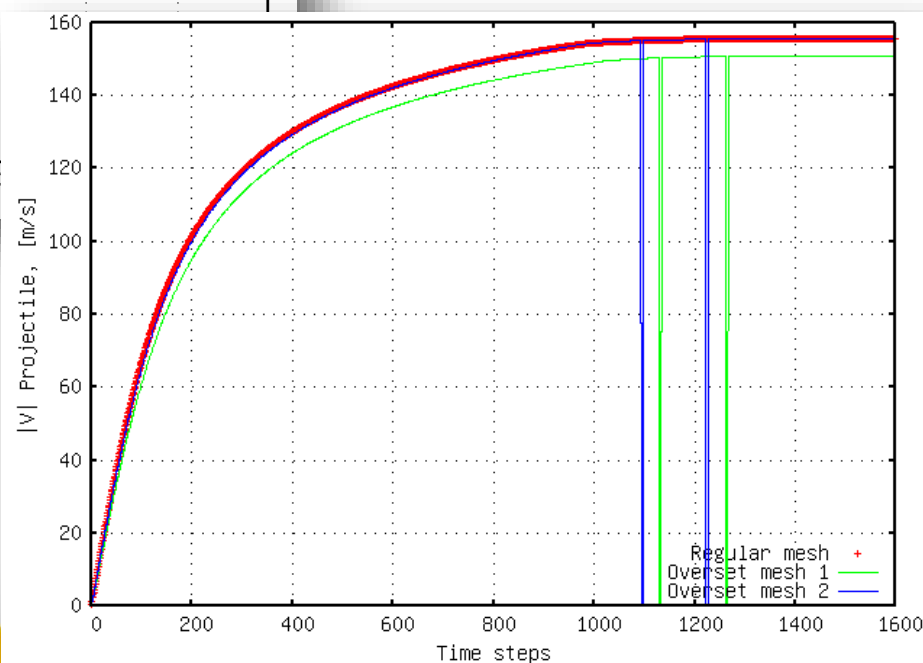


# Case3 : Result

## Projectile position history



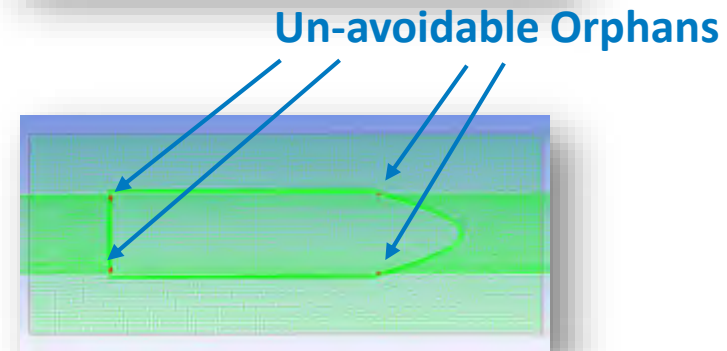
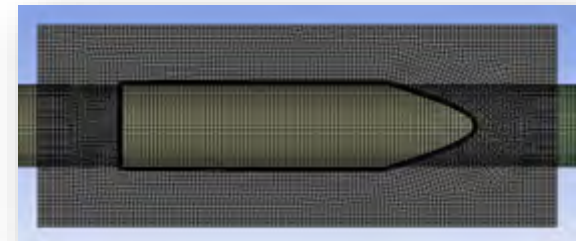
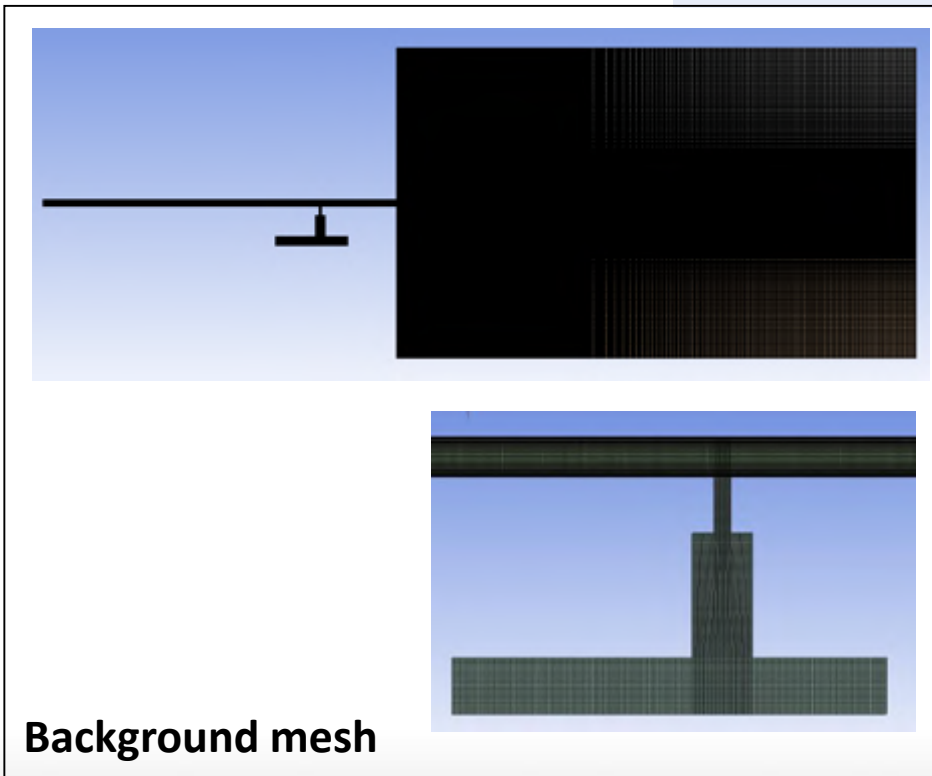
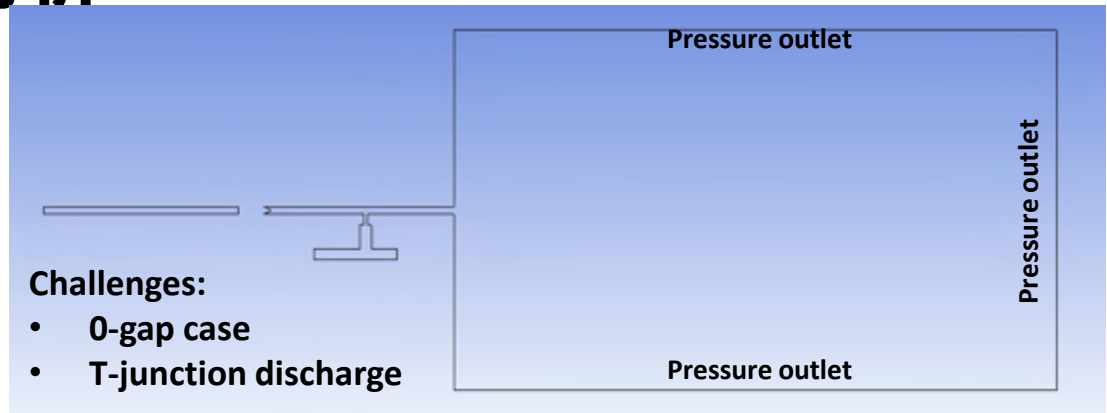
## Projectile velocity history



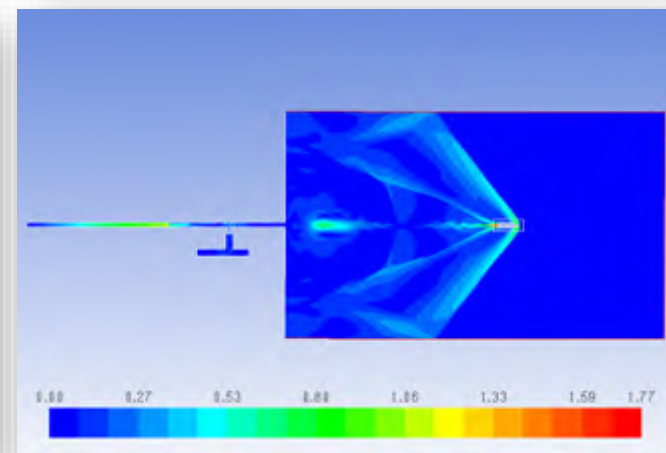
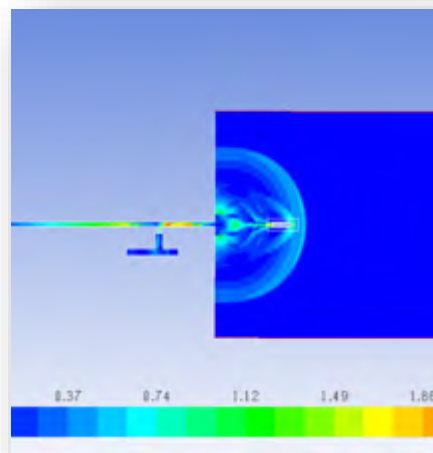
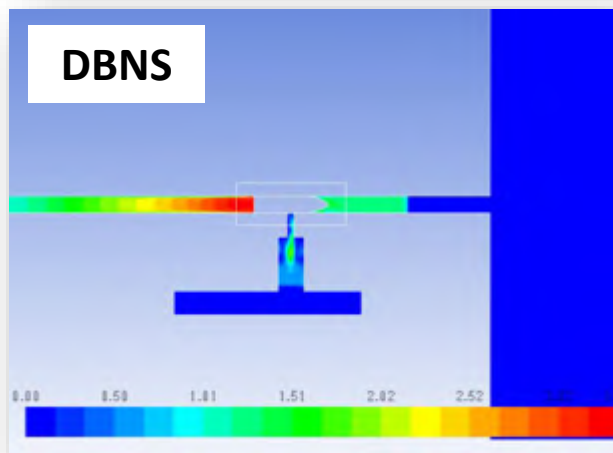
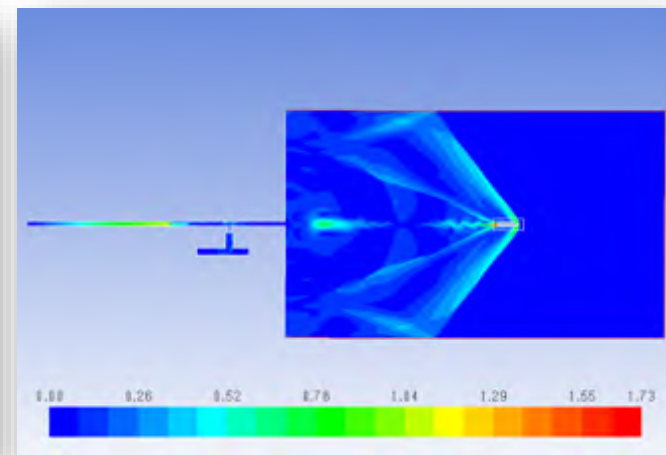
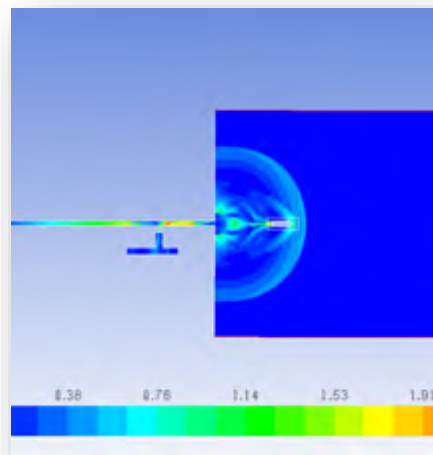
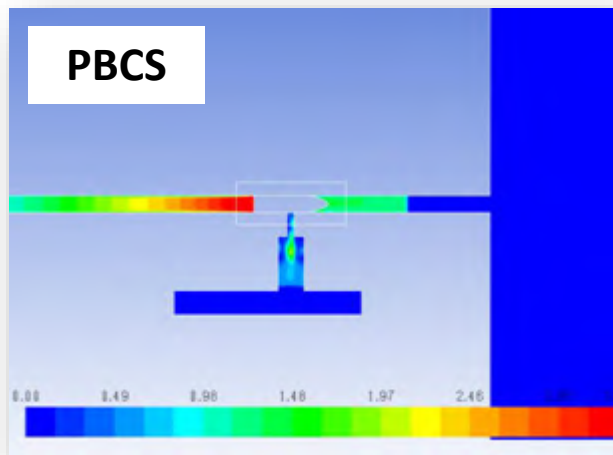
- Orphans , dead cells和弹丸后壁面影响弹丸的驱动力
- 由于在muzzle breaks处存在orphans和dead cell导致的后处理问题

# Case4 : 2D弹头运动分析

- 网格 & 设置
- PBNS & DBNS
- SST k-omega
- Noble-Abel real gas

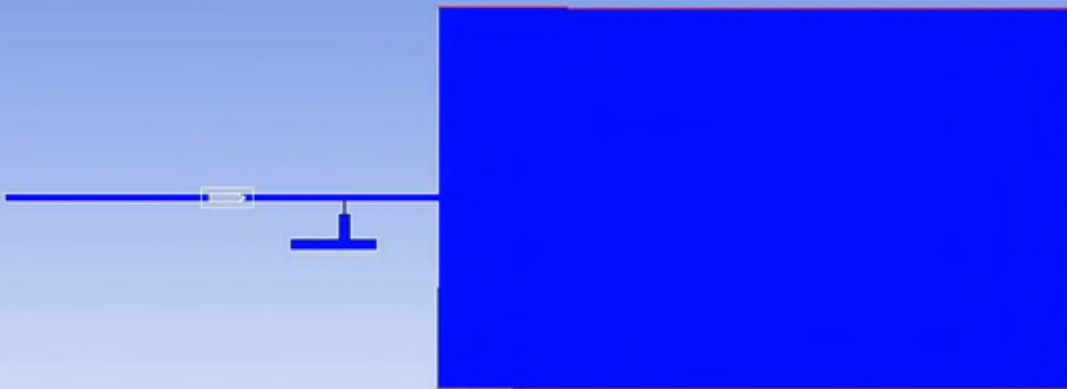


# Case4 : Result, 马赫数



# Case4 : Result, 马赫数

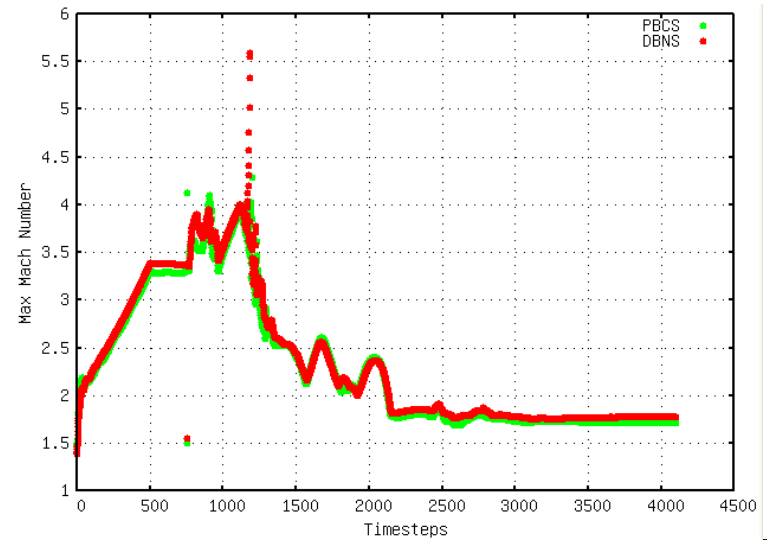
DBNS



0.00 0.22 0.44 0.66 0.88 1.10 1.32 1.47



Contours of Mach Number (Time=1.0000e-06) Jan 12, 2017  
 ANSYS Fluent (2d, dp, dbns imp, overset mesh, sstk, transient)

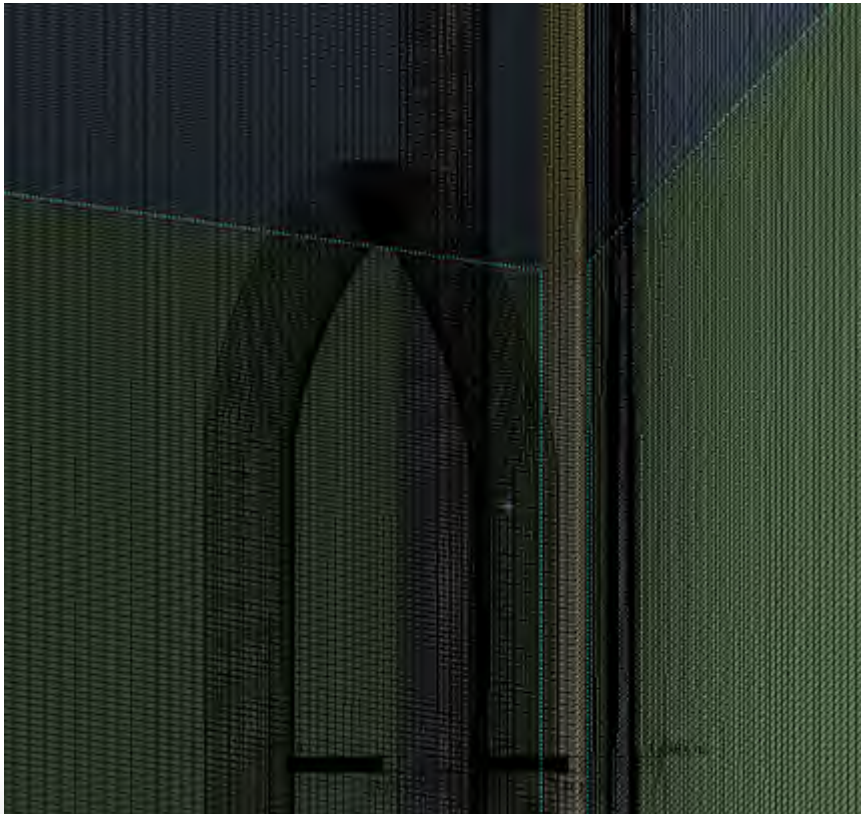


# Case5 : ULA Stage Separation

Free Stream Mach = 3.0 (Note: DBNS is not supported for Overset Mesh at R17.0)

## Overset Mesh

Quarter symmetry, hexahedral mesh with inflation, **1.67M cells**



## MDM Mesh

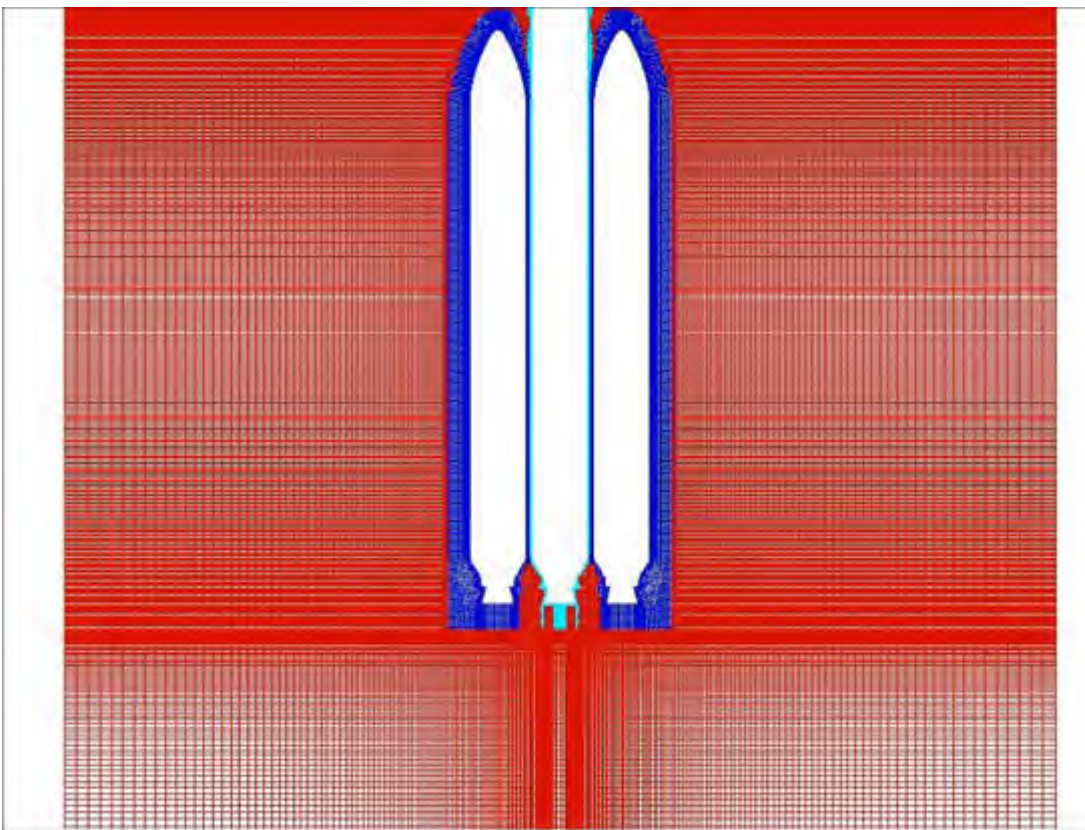
Quarter symmetry, hybrid mesh with inflation, **6M cells**



Remeshed zone

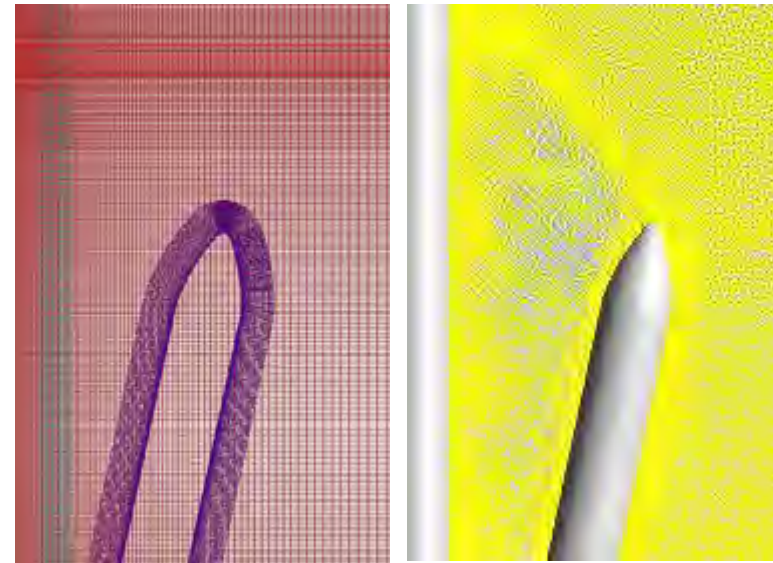


# Case5 : ULA Stage Separation



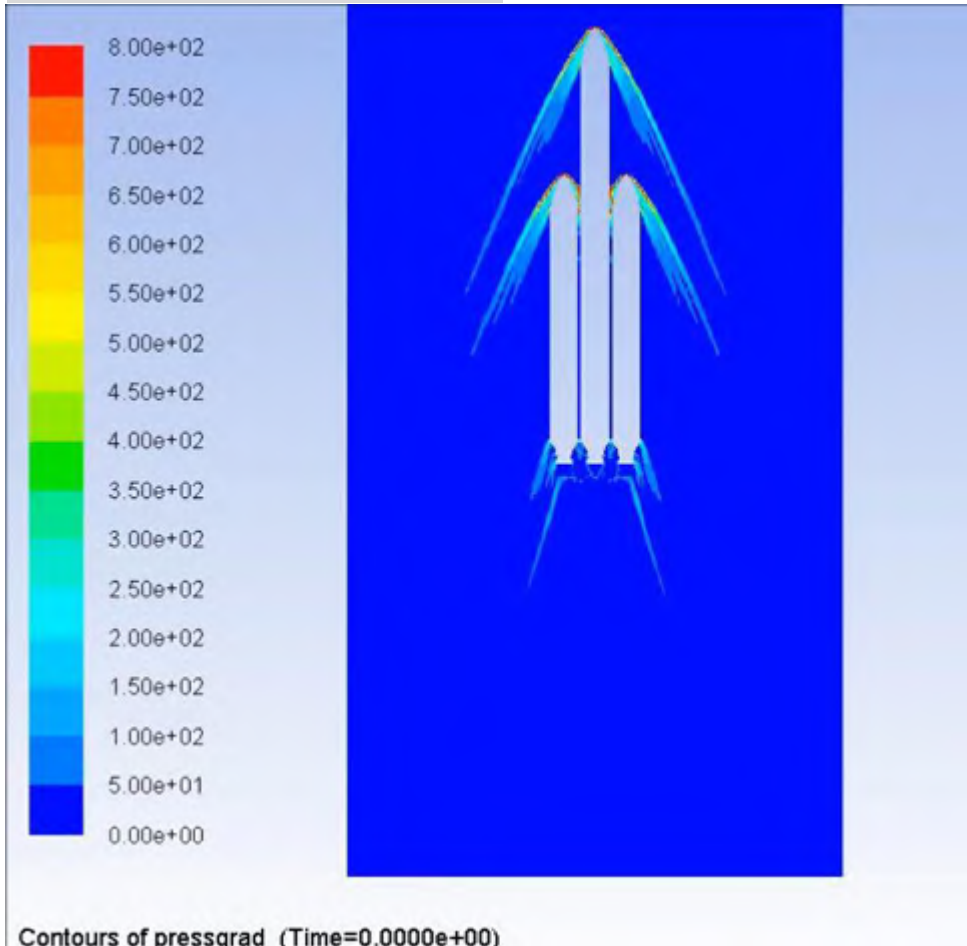
Mesh (Time=0.0000e+00)

**Mesh at 1.0s:**  
**Original mesh quality preserved in**  
**overset mesh**

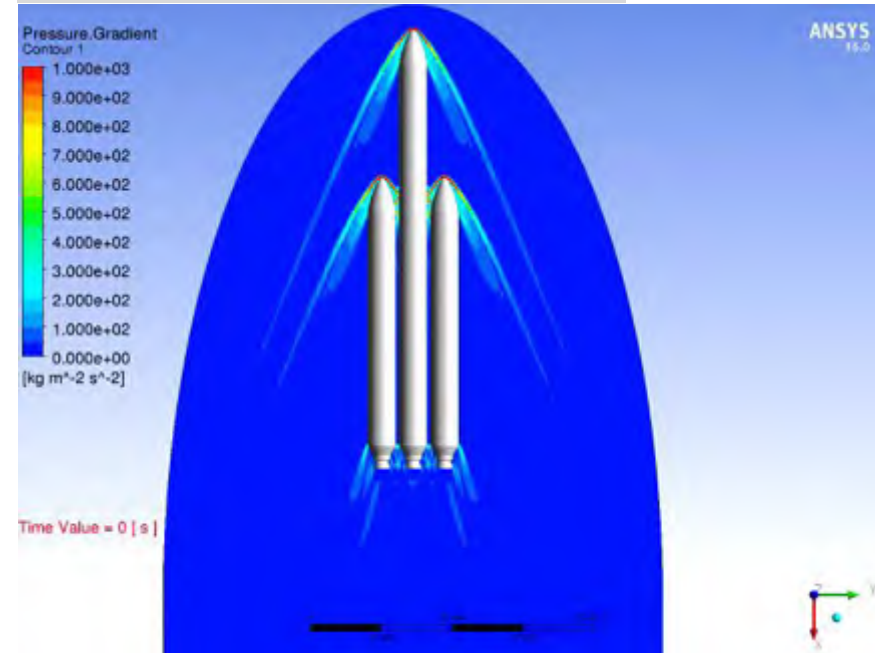


# Case5 : ULA Stage Separation

## Overset Mesh Solution



## MDM (Remeshing) Solution



- **No Mesh Degradation in Overset Mesh and it preserves quality of Shock Capture**

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